

NAVAL POSTGRADUATE SCHOOL

Monterey, California



THESIS

GROUND-UP-TO-SPACE (GUTS)
LASER PROPGATION CODE
DESCRIPTION AND MANUAL

by

Joel Steven Morrow

June 1984

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Approved for public release; distribution unlimited Prepared for: Dr. Fred Raymond, Code 9110

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This thesis prepared in conjunction with research supported in part by the Naval Research Laboratory, Washington, D.C.

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SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM								
NPS 67-84-008	3. RECIPIENT'S CATALOG NUMBER								
Ground-Up-to-Space (GUTS) Laser Propagation Code	S. TYPE OF REPORT & PERIOD COVERED Master's Thesis June 1984								
Description and Manual	6. PERFORMING ORG. REPORT NUMBER								
7. AUTHOR(s)	8. CONTRACT OR GRANT NUMBER(#)								
Joel Steven Morrow	N0001484WRP4012								
Naval Postgraduate School Monterey, California, 93943	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS								
11. CONTROLLING OFFICE NAME AND ADDRESS Naval Postgraduate School	June 1984								
Monterey, California, 93943	13. NUMBER OF PAGES 141								
14. MONITORING AGENCY NAME & ADDRESS(It ditterent from Controlling Office)	15. SECURITY CLASS. (of this report)								
Dr. Fred Raymond Naval Research Laboratory	Unclassified								
4555 Overlook Ave. SW Washington, D.C. 20375	154. DECLASSIFICATION/DOWNGRADING SCHEDULE								
Approved for public release; distribution	on unlimited								
17. DISTRIBUTION STATEMENT (of the ebstract entered in Block 20, if different from	n Report)								
18. SUPPLEMENTARY NOTES									
19. KEY WORDS (Continue on reverse elde it necessary and identify by block number) High Energy Laser									
Laser Propagation									

20. ABSTRACT (Continue on reverse elde if necessary and identify by block number)

Laser Weapon Simulation Program

ASAT Laser Simulation

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SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

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Laser Propagation Code
Description and Manual

by

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Lieutenant, United States Navy
B.S., University of South Carolina, 1976

Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN AERONAUTICAL ENGINEERING

from the

NAVAL POSTGRADUATE SCHOOL June 1984

Thesis 188397

AESTRACT

GITSAVG is a high energy laser propagation computer program for ground-to-space applications. Written by Dr. C. E. Hogge from the Air Force Weapons Laboratory, Kirtland AFB, it is one in a family of propagation codes addressing this application. Specifically, GUTSAVG was designed to compute irradiance at the target given a model atmosphere, laser device parameters, and simple target engagement geometry. The transmitter induced effects of beam quality and jitter are considered as are the linear atmospheric effects of scattering, absorption, and turbulence. A thermal blooming model is also included. Adaptive optics compensation can be applied with consideration given to isoplanatic effects.

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I. INTRODUCTION

GCTSAVG is a simplified laser propagation program. It is intended specifically for near vertical ground-to-space applications utilizing a fixed earth-based transmitter directed at a single target satellite. The primary purpose of the program is to provide irradiance and fluence on target along with related propagation data.

The program was written by Charles B. Hogge from the Air Force Weapons Laboratory, Kirtland Air Force Base, Albuquerque, New Mexico. GUTSAVG is one of a family of ground-to-space propagation programs. Other versions include GUTSFF (footprint) and GUTSMTF. GUTSFP computes an engagement envelope based on user supplied irradiance threshold levels. The propagation calculation methods are identical to GUTSAVG. GUTSMTF is a full wave-optics program using fast Fourier transforms in the beam propagation computations.

The hasic approach used in GUTSAVG is to utilize modulation transfer functions to characterize effects such as beam quality, jitter, turbulence and to apply these effects at the aperture as a single phase screen. The same approach is used in FSF-IV [Ref. 1]. The general modulation transfer function, (MTF), used in this way is described in following sections along with the development of each beam degrading mechanism. A linearized model is used for thermal blooming which is also effectively applied at the aperture as a phase screen. In the case of blooming, however, the phase variance due to blooming is computed, and the Strehl relation is used to determine the relative irradiance reduction. The effect of thermal blooming and the other spreading effects are combined using both FSS and multiplicative methods. Finally, the average of these two methods is used to determine the

total system irradiance reduction from the diffraction limited case.

II. PROPAGATION FEATURES

A. IEE MIF APPROACH

GETSAVG uses MTFs to apply and then to consolidate the effects of jitter, bear quality, and turbulence. Essentially, these effects are replaced by a phase screen at the aperture which multiplies the initial complex aperture distribution by a random phase distortion factor.

Iutcmirski and Yura [Ref. 2] have developed an expression for the average intensity at a point P in the far-field due to perturbations at the aperture. Ine intensity at P is

$$I(\overline{p}) = \left(\frac{k}{2\pi z}\right)^{2} \int_{\Phi} M_{\phi}(\overline{p}, z) \exp\left[-\left(\frac{i \cdot k}{z}\right) \overline{p} \overline{\rho}\right] d\overline{\rho} d\overline{\rho} + \int_{\Phi} U(r + \frac{1}{2}\overline{\rho}) U^{*}(r - \frac{1}{2}\overline{\rho}) \exp\left[-\left(\frac{i \cdot k}{z}\right) r \overline{\rho}\right] d\overline{r} d\overline{r}$$
(2.1)

where

$$\overline{\rho} = |\mathbf{r}_2 - \mathbf{r}_1| \tag{2.2}$$

and

$$r = \frac{r_2 + r_1}{2}$$
 (2.3)

 \overline{F} is the vector from the z axis of symmetry in the far-field to the point P. U(r) is the aperture distribution and M $_{\varphi}$ ($\overline{\rho}$,z) is the MTF for the disturbance. r_1 and r_2 are two arbitrary points in the aperture where the phase perturbation is measured. $\overline{\rho}$ is the distance between the points. z, in the

case considered here, is a constant. If the phase disturlance is a random Gaussian variable with a known correlation function, then the MTF can be expressed as

$$M_{\phi}(\bar{\rho}) = \langle \exp[i(r_1 - r_2)] \rangle \qquad (2.4)$$

In terms of the structure function,

$$M_{\phi}(\overline{p}) = \exp\left[(-\frac{1}{2})D_{\phi}(r_1 - r_2)\right]$$
 (2.5)

The second integral in equation 2.1 represents the urnormalized aperture MTF. Normalizing this term with the power in the aperture, (P_0) , to cause the MTF to be unity at the crigin results in

$$I(\overline{p}) = \left(\frac{k}{2\pi z}\right)^2 P_0 \int M_{\phi}(\overline{\rho}) M_{a}(\overline{\rho}) e^{-\left[ik\overline{\rho}\overline{p}\right]} d^{\frac{2}{\rho}}$$
(2.6)

Noting the symmetry of the intensity for a given p and expressing $M_{\varphi}(\rho)$ as the combined effects of jitter, rear quality, and turbulence, equation 2.6 can be rewritten as

$$I(\overline{p}) = \left(\frac{k}{2\pi z}\right) P_0 \int M_{j}(\overline{p}) M_{t}(\overline{p}) M_{b}(\overline{p}) M_{a}(\overline{p}) J_0(\frac{k\overline{p}\overline{p}}{z}) \overline{p} d\overline{p}$$
(2.7)

A Fourier-Eessel transform has been used. Note that the MIFs cf jitter (M_j), beam quality(M_b), and turbulence(M_t) have been substituted for \mathcal{E}_{φ} .

The FSF-IV manual [Ref. 3] contains the development above in more detail. MIFs for the specific effects are described in following sections.

E. TEEREAL BLCOMING

The thermal blooming model in GUTSAVG is based on the following linearized density perturbation equation.

$$\frac{\Delta \rho}{\rho_0} = \frac{\alpha(\ell)}{V_0} \frac{\gamma - 1}{\gamma} \frac{1}{P_0} \int_{-\infty}^{\infty} I(x', y) dx' \times \exp\left[-\int [\alpha(\ell) + \sigma(\ell)] d\ell\right]$$

Here, $\alpha(\ell)$ is the atmospheric absorption coefficient at a distance ℓ along the leam, V_0 is a constant transverse wind velocity, and p_0 is the ambient pressure. The exponential term represents the total extinction due to scattering and absorption. x'is a constant of integration. It is assumed that the beam is propagated in the positive z direction a distance ℓ and that the wind vector is in the positive x direction. The intensity integral represents the heating of the atmosphere as it transits the beam [Ref. 4]. Some assumptions embodied in the above equation are that $V_0 << c_0$, the local sonic velocity, so that the process represented cocurs at constant pressure, and that the kinetics of absorption and conversion to heat are extremely fast [Ref. 5].

Ey applying the Gladstone-Tale relation, the density relation can be expressed as a change in the refractive index.

$$\Delta n = (n_0 - 1) \frac{\Delta \rho}{\rho_0}$$
 (2.9)

Equation 2.8 can be rewritten as

$$\Delta n = (n_0 - 1) \frac{\alpha(\ell)}{V_0} \frac{\gamma - 1}{\gamma} \frac{1}{P_0} \int_{\infty}^{X'} I(x', y) dx'$$

$$\times \exp\left[-\int [\alpha(\ell) + \sigma(\ell)] d\ell\right]$$
(2.10)

The change in the wavefront phase of a beam due to the refractive index charge when the beam is propagated a distance g, is given by

$$\Delta \phi = \frac{2\pi}{\lambda} \int_0^{\ell} \Delta n d\ell \qquad (2.11)$$

Substituting equation 2.10 into 2.11 and changing the limits of integration to reflect the ground-to-space propagation path results in _______x'

Alts in
$$\Delta \phi(\mathbf{x}, \mathbf{y}) = \left[\frac{2\pi}{\lambda} \frac{\mathbf{n}_0 - 1}{\mathbf{p}_0} \frac{\mathbf{y} - 1}{\mathbf{y}} \int_{-\infty}^{\mathbf{x}'} \mathbf{I}(\mathbf{x}', \mathbf{y}) d\mathbf{x}' \right]$$

$$\times \int_{\mathbf{h}_1}^{\mathbf{h}_{atm}} \frac{\mathbf{n}_0 - 1}{\mathbf{p}_0} \frac{\mathbf{y} - 1}{\mathbf{y}} \int_{-\infty}^{\mathbf{x}'} \mathbf{I}(\mathbf{x}', \mathbf{y}) d\mathbf{x}' \right]$$

$$\times \int_{\mathbf{h}_1}^{\mathbf{h}_{atm}} \frac{\mathbf{n}_0 - 1}{\mathbf{p}_0} \frac{\mathbf{y} - 1}{\mathbf{y}} \int_{-\infty}^{\mathbf{x}'} \mathbf{I}(\mathbf{x}', \mathbf{y}) d\mathbf{x}'$$

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$$\times \int_{\mathbf{h}_1}^{\mathbf{h}_{atm}} \frac{\mathbf{n}_0 - 1}{\mathbf{p}_0} \frac{\mathbf{y} - 1}{\mathbf{y}} \int_{-\infty}^{\mathbf{h}_{atm}} \mathbf{I}(\mathbf{x}', \mathbf{y}) d\mathbf{x}'$$

$$\times \int_{\mathbf{h}_1}^{\mathbf{h}_{atm}} \frac{\mathbf{n}_0 - 1}{\mathbf{p}_0} \frac{\mathbf{n}_0 - 1}{\mathbf{y}} \int_{-\infty}^{\mathbf{h}_{atm}} \mathbf{n}_0 d\mathbf{n} d\mathbf{n}$$

 h_{atm} is the extent of the atmosphere, which is about about 30 km, and h_t is the height of the laser transmitter. Also, the wind term has been expanded to include the relative wind velocity due to slewing. V₀ is assumed to be parallel and opposite in direction of that of the target motion. ξ is the angle of incidence of the wind to the beam so that V₀ccs(ξ) represents the transverse wind. wh is the effective wind generated by slewing. ω is the angular slew rate.

The first portion of equation 2.12 is independent of path while the second part is not, assuming I(x,y) does not change along the progragation path. This assumption is valid only for a very small amount of blooming. The approach used

in GUTSAVG is to determine the phase distortion lue to thermal blooming by first evaluating the path invariant part of equation 2.12. This is accomplished by constructing a phase screen at the aperture and then removing the best fit tilt, focus curvature, and mean phase. Zernike polynomials are used to model these aberrations. The result is the residual phase due to thermal blooming alone. The variance of the phase is then computed.

The path dependent term is evaluated within the angle interval loop of the program and is applied to the previously computed phase variance during each path iteration. The path iteration process is diagrammed in the engagement geometry section. Also, see Figures 2.1 and 2.2 for a flow diagram of the general treatment of thermal blooming in the program.

Once the total phase variance has been determined, the Strehl relation is used to compute the intensity degradation due to thermal blooming.

$$\frac{I}{I_0} = \exp(-\sigma^2) \tag{2.13}$$

The result of equation 2.13 is a relative intensity (Irel) ratio. Io is the ideal on-axis irradiance with no phase distortion. The Strehl relation above is thought to be too severe a model for Irel below 0.3 [Ref. 6]. For that reason, if σ^2 is less than 1.2, the Irel will be computed using polynomial curve fits developed from GUTSMIF results. GUTSMIF is a full wave optics code utilizing fast Fourier transforms. For a description of the curve fit method above, see the subroutine BLOOM explanation. [Ref. 7]

Combining the thermal blooming effect with the other effects, such as turbulence and jitter, is accomplished by averaging the results of two different approaches. The first

appreach is the RSS (roct sum squared) method. This method of contining the Irel due to the effects of thermal blooming with the Irel due to jitter, beam quality, and turbulence is accomplished as follows

$$Irel_{rss} = \left(1 + \left[\frac{1}{Irel_{tb}} - 1\right] + \left[\frac{1}{Irel_{o}} - 1\right]\right)^{-1}$$
 (2.14)

where Irel is the thermal blocming result and Irel is the result of the other effects. The second approach is a multiplicative approach and is simply

$$Irel_m = Irel_o \times Irel_{tb}$$
 (2.15)

The two contined Irels obtained by these methods are then averaged to give the total intensity ratio due to all the attenuating or distorting propagation effects.

$$\frac{I}{I_0} = Irel_{tot} = \frac{Irel_m + Irel_{rs}}{2}$$
 (2.16)

Io is the ideal diffraction limited on-axis intensity. The bases for the above averaging process is empirical in nature and is an attempt to adjust the results obtained by the RSS method alone. The results produced by RSS were thought to be too optimistic. The multiplicative method, a more pessimistic approach, was therefore included. The ultimate Irels obtained are very close to those obtained by GUTSMTF, the full wave code. [Ref. 8]

There are some limitations to the thermal blooming model used in GUTSAVG in addition to the assumptions already mentioned. First, the wind (V_0) is applied as a constant

everywhere in the atmosphere. At higher altitudes, this is nct a major consideration due to the higher relative velocity of the beam. At low to medium altitudes this could affect thermal blooming to an extent to warrant the addition cf a wind profile as a function of altitude. This could be done with little effort if the data is available. Also, the rrogram assumes a wind parallel and opposite to the direction of the target motion. This precludes the case of slewing with the wind and the creation of null spots. Transcric blooming, which viclates one of the original assumptions of the equation used, is not considered [Ref. 9]. For very low altitude satellites, the high slew rates generated would result in a supersonic relative wind across the leam. In this case, the constant pressure assumption is invalid [Ref. 10]. Kinetic cooling and molecular treakdowr are also not addressed in GUTSAVG.

C. SCATTERING AND AESORPTICN

with respect to scattering and absorption data. Extinction coefficients must be entered by the user in the appropriate subroutines, ALFS and ALFA, or the program can be modified to accept a separate data file. Once inserted into the program, scattering and absorption coefficients are treated without distinction between aerosol and molecular mechanisms. Therefore, the coefficients used must represent the total effect of scattering or absorption.

$$\alpha(h) = \alpha(h)_{mol} + \alpha(h)_{aer}$$
 (2.17)

$$\sigma(h) = \sigma(h)_{mol} + \sigma(h)_{aer}$$
 (2.18)

Transmission due to scattering and absorption are computed indentically and given by

$$T_{a} = \exp\left(-\sec\theta \int_{h_{+}}^{h_{a} tm} \alpha(h) dh\right) \qquad (2.19)$$

where θ is the zenith angle, and the integration limits are the altitude of the transmitter and the vertical extent of the atmosphere. The extent of the atmosphere in the program is defined as 30 km. Figure 2.3 shows the program application of scattering and absorption.

C. FFAM CUALITY

aperture in terms of the electromagnetic field amplitude distribution and total beam quality. The intital amplitude field used by the program is Gaussian in shape with the user specifying a waist diameter. The waist diameter is defined by the 1/e² point on the distribution. Truncation of the this Gaussian field will of course depend on the aperture diameter and the size of the central obscuration. If the waist diameter is made large compared to the aperture, a more uniform distribution results.

Exam quality at the aperture exit may be specified by two different parameters. One of these is the 'times diffraction limited number', N. N has been used in a general way to mean an increase in far-field spot size or as a 'power-in-the-bucket' ratio. In GUTAVG, N is a total heam quality term. The second beam quality parameter is a nondimensional term representing the RMS phase distortion at the laser wavelength at the exit aperture, $\frac{\delta_{\rm rms}}{\rm rms}$.

The parameters are related to each other and to the intensity degradation by

$$\frac{I}{I_0} = \frac{1}{N^2} = \exp\left(-\left(\frac{2\pi\delta_{rms}}{\lambda}\right)^2\right)$$
 (2.20)

To apply beam quality to the propagation problem, at MTF array is constructed representing a phase screen at the aperture. The phase is assumed to be a Gaussian random variable with a zero mean value. For the axi-symmetric heam considered in GUTAVG, the MTF is [Ref. 11]

$$M_{b}(\overline{\rho}) = \exp\left(-k^{2}\left[\sigma^{2} - C_{\phi}(\overline{\rho})\right]\right)$$
 (2.21)

C is the autocorrelation function of the phase and is defined as

$$C_{\phi}(\overline{\rho}) = \sigma^2 \exp\left[-\left(\frac{\overline{\rho}}{L}\right)^2\right]$$
 (2.22)

where σ^2 is the phase variance, $(\frac{2\pi\delta_{rms}}{\lambda})^2$, and L is the phase correlation length [Ref. 12]. The beam quality MIF array is combined with the MTF arrays due to other propagation effects to determine the complete system MTF and, hence, the total irradiance degradation. Based on the provided input parameter, Figure 2.4 shows the general treatment of beam quality within the program.

F. TUBEULENCE

GCTSAVG uses the Eufnagel model [Ref. 13] for C_n^2 as an indicator of the optical effects of turbulence along the propagation path. C_n^2 is the refractive index structure constant and represents the refractive index in the

atmosphere as a function of turbulence induced density fluctuations. The model is an empirically derived vertical profile of C_n^2 .

Fried [Ref. 14] has developed a parameter which is directly related to the behavior of a coherent beam in a turbulent medium. This term is called the effective coherence diameter, r. In the case of a laser transmiter, r. represents a physical limit to the transmitter diameter of a near diffraction-limited beam. For a transmitter diameter, D. larger than r., degradation of the beam by turbulence will occur. If D is smaller than r., then near diffraction-limited propagation will be achieved. Beam wander or pure tilt occurs for transmitter diameters approximately equal to r. Yura [Ref. 15] has defined a somewhat different but related term that can be thought of as a lateral coherence length, p. These two quantities are given by

$$r_0 = \left[\frac{2.19}{6.88} k^2 \sec \theta \int C_n^2(h) dh\right]^{-3/5}$$
 (2.23)

and

$$\rho_0 = \left[1.45 \text{ k}^2 \sec \theta \int C_n^2(h) dh\right]^{-3/5}$$
 (2.24)

so that

$$\rho_0 = \frac{r_0}{2 \cdot 1} \tag{2.25}$$

where $k=\frac{2\pi}{\lambda}$, θ is the zenith angle, and the limits of integration are h , the vertical extent of the atmosphere, and lt ,the height of the transmitter.

the program may be allowed to compute it based on equation 2.24. The user may also input ρ_0 indirectly by specifying the 'seeing conditions', a quantity used by astronomers to describe angular spread of a stellar point source (see input section and Figure 2.5). The program uses ρ_0 to compute the atmospheric MIF.

The NTF of the turbulent atmosphere is determined by developing the structure function of the turbulence. This development is demonstrated by Yura [Ref. 16]. The resulting MTF is given by

$$M_{t}(\overline{\rho}) = \exp\left[-\left(\frac{\overline{\rho}}{\rho_{0}}\right)^{3/5}\right]$$
 (2.26)

This MTF effectively applies the the effect of tirbulence along the propagation path as a phase screen at the aperture.

F. JITTER

Beam jitter is a user input and is specified by the variance of the angular excursions of the beam. Jsing the 2-sigma-p definition,

$$2\sigma_{p} = \sqrt{2(\sigma_{x}^{2} + \sigma_{y}^{2})}$$
 (2.27)

where $\sigma_{\rm X}$ and $\sigma_{\rm Y}$ are the axial variances of the jittered beam center in the far-field. $\sigma_{\rm X}$ and $\sigma_{\rm Y}$ are random variables with Gaussian distribution and in the symmetric case , as considered by GUTSAVG, $\sigma_{\rm X}^{}=\sigma_{\rm Y}^{}$. The resultant intensity distribution due to jitter will also be a Gaussian distribution with ($2\sigma_{\rm p}^{}$) representing the spot radius defined at the 1/e² point. In other words, 86.5% of the beam energy will reside within the radius $2\sigma_{\rm p}^{}$. [Ref. 17]

Jitter can be shown to be a wavefront tilt at the aperture. Using this approach, a phase screen at the aperture can be used to characterize the effects of jitter and a jitter MIF developed. That MTF is given by

$$M_{j}(\overline{\rho}) = \exp\left[-\frac{k^{2}\overline{\rho}^{2}(2\sigma_{p})^{2}}{8}\right] \qquad (2.28)$$

Figure 2.6 is a general flow diagram for the treatment of jitter within the program.

G. ALAPTIVE OPTICS

The ability to apply adaptive optics corrections to the propagation problem has been included in GUTSAVG. The user has several options with respect to the type and degree of compensation desired. The following general discussion and figure 2.8 provides the needed insight to the effects of selecting the adaptive optics options.

The major compensation mode provided by the program is invoked by selecting full zonal adaptive optics with consideration given to isoplanatic effects. When selected, this model results in the correction of beam degradation due to turbulence. This is accomplished by correcting turbulence induced tilt and then adjusting ρ_0 so as to produce a predetermined level of adaptive optics performance. This predetermined performance is as measured by the Strehl ratio qiven a residual phase variance determined by the adaptive cytics sensor phase. Parameters determining the phase error are the response bandwidth of the adaptive optics system, the number of system actuaters, the reflected radiant intensity of the target, and the target-to-sensor transmission. The resultant Po found in this manner is then used to compute the atmospheric MTF. When this 'corrected' MTF is used to compute the far-field intensity, the result will represent an adaptive optics corrected value.

without invoking full zonal adaptive optics, the user may apply a tilt-only correction for turbulence. In this case some or all of the tilt due to turbulence may be removed before computing the atmospheric MTF. The degree of tilt compensation is specified by the user.

As mentioned above, isoplaratic effects are included in the adaptive optics calculations. This is also an option, however, and isoplanatic calculations may be inhibited by the user. The effect considered is the limitation of the adaptive optics system given an isoplanatic angle smaller than the target lead angle. Fried [Ref. 18] provides a discussion of isoplanatism and development of the isoplanatic argle.

Although adaptive optics compensation for thermal blocking is not modeled in a strict sense, a thermal blocking correction factor can be applied. This factor is simply a fractional constant that multiplies the thermal blocking phase variance before the Strehl relation is used to compute the intensity degradation.

Refer to the input definitions and the subroutines involved with adaptive optics. In addition, refer to figure 2.7 for more explanation.

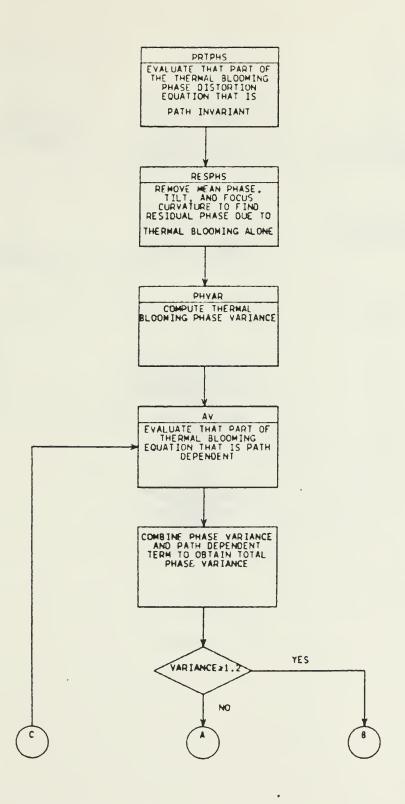


Figure 2.1 GLISAVG Thermal Blooming Algorithm.

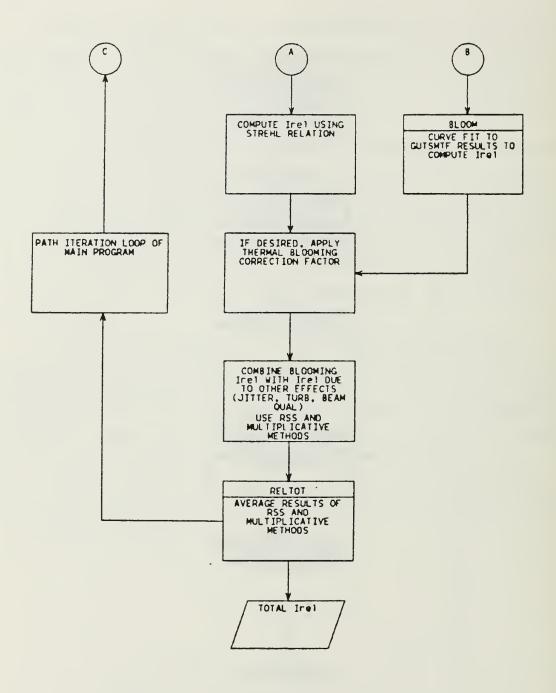


Figure 2.2 Thermal Electing Algorithm (cont).

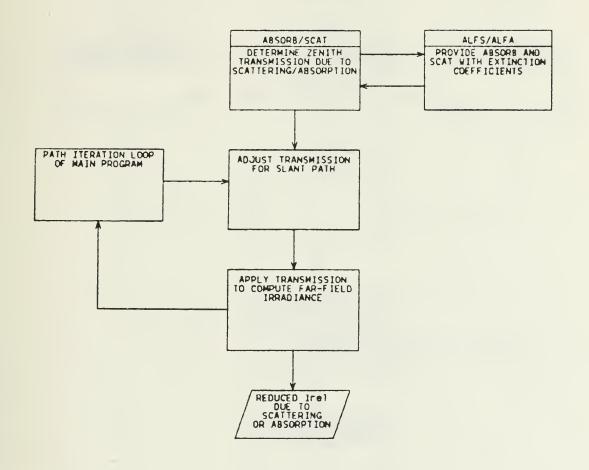


Figure 2.3 GUISAVG Scattering and Absorption Algorithm.

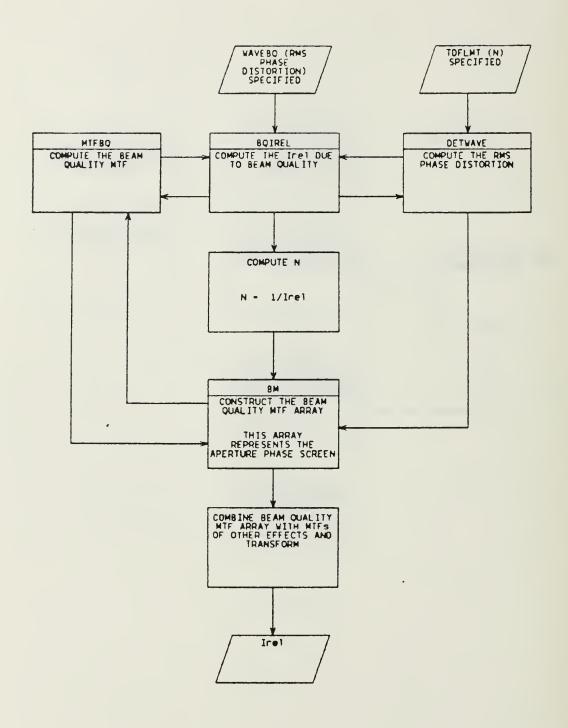


Figure 2.4 GUISAVG Beam Quality Algorithm.

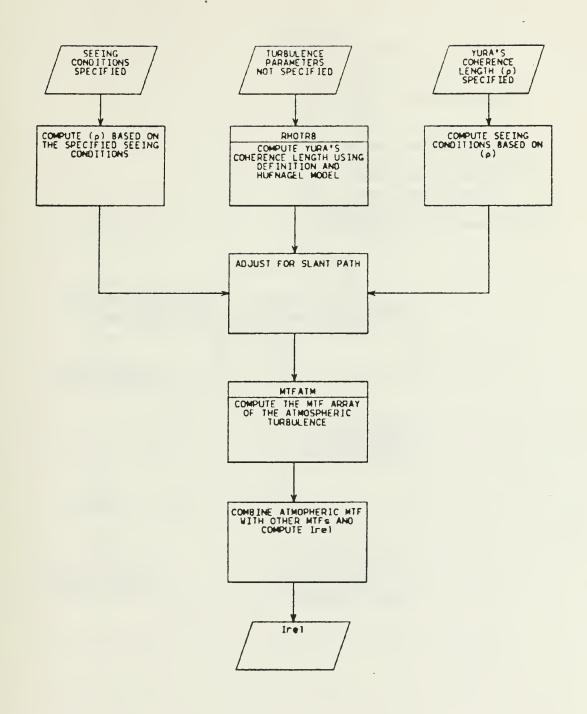


Figure 2.5 GUTSAVG Turbulence Algorithm.

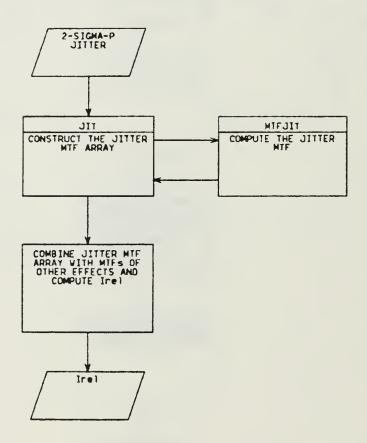


Figure 2.6 GUTSAVG Jitter Algorithm.

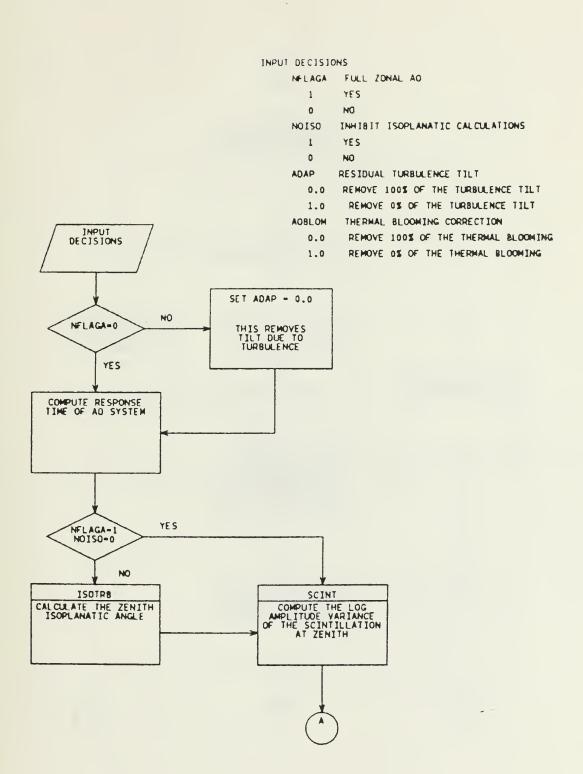


Figure 2.7 GUISAVG Adaptive optics Algorithm.

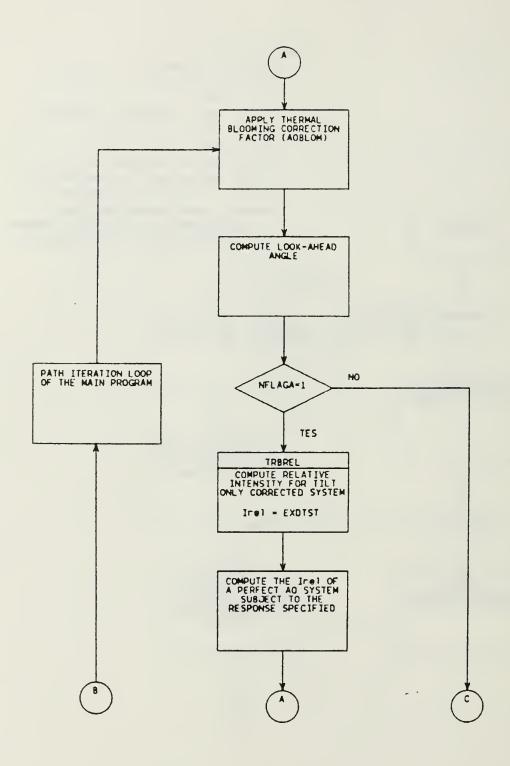


Figure 2.8 GUTSAVG Adaptive Optics Algorithm (ccnt).

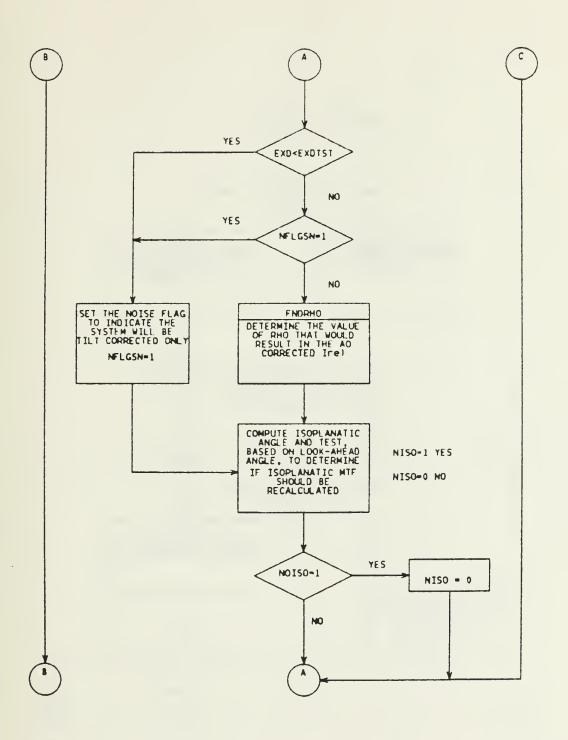


Figure 2.9 GUTSAVG Adaptive Optics Algorithm (ccnt).

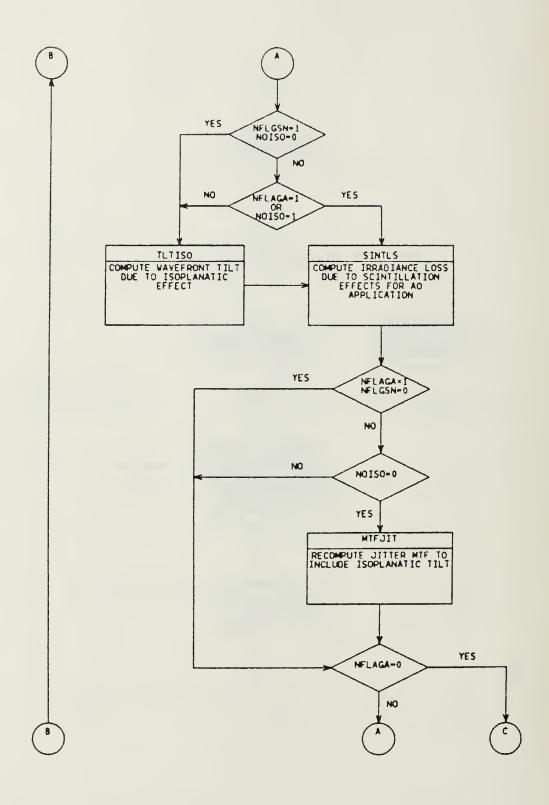


Figure 2.10 GUTSAVG Adaptive Optics Algorithm (ccnt).

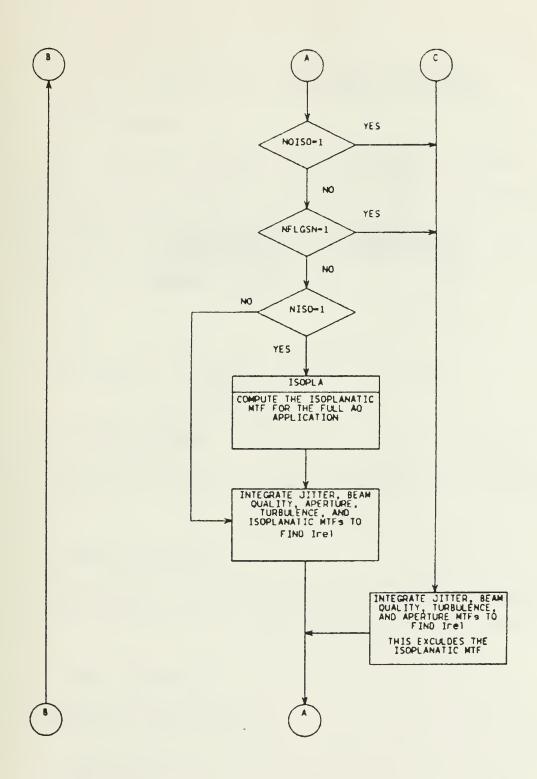


Figure 2.11 GUTSAVG Adaptive Optics Algorithm (ccnt).

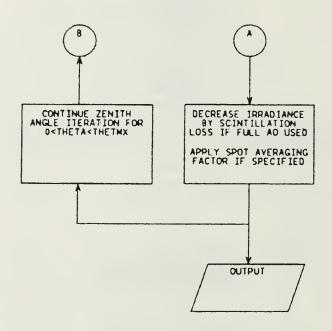


Figure 2..12 GUTSAVG Adaptive Optics Algorithm (ccnt).

III. PROGRAM AND SUBFROGRAM DESCRIPTION

A. FECGEAR INPUTS

The following are the inputs necessary to utilize GUTSAVG. Default values are indicated for parameters not absolutely required for program operation. At NPS these input parameters are entered via an input file. A copy of this file is provided in Appendix A. (*) indicates a nondimensional parameter.

* CIA (meters)

The diageter of the transmitter aperture.

• IIACES (meters)

The diameter of the central obscuration of the transmitter.

• FFAMS2 (meters)

The Gaussian waist diameter of the amplitude distribution at the aperture. (measured at the $1/e^2$ point)

• WAVE (meters)

The laser wavelergth.

• FICIAL (Watts)

Ite total power at the aperture.

• IDFIMI (*)

This is the ofter used "times diffraction limited number" and represents total heam quality. As used in GUTS, it is related to the RMS phase distortion at the aperture by the Strehl approximation

$$\frac{1}{(\text{TDFLMT})^2} = \exp\left[-\left(\frac{2\pi\delta_{\text{rms}}}{\lambda}\right)^2\right]$$
 (3.1)

(TDFIMT)^2 , therefore, is equivalent to the ratio of the cr-axis diffraction limited intensity to the on-axis intensity resulting from the near-field phase distortion, $\delta_{\rm rms}$.

$$\frac{1}{(\text{TDFLMT})^2} = \frac{I}{I_0}$$
 (3.2)

For a discussion or the limitation of 3.1, see [Ref. 19].

· WAVEEC (*)

This term is the FMS phase distortion at the laser aperture, nondimensicalized by the wavelength.

$$WAVEBQ = \frac{\delta_{rms}}{\lambda}$$
 (3.3)

• SCALEÇ (meters) default = DIA/5

This term is the transverse phase correlation length at the aperture.

• THSEE (arcsec) default = f(RHOO)

This is a qualitative term used by astronomers to describe 'seeing conditions' in the visible range. If a point source is viewed from the earth, it may not appear as a point source but as a 'smear' or spot. The angular spread of this spot is the parameter THSEE. If RHOO is not specified as an input and THSEE is, the program will use THSEE to compute RHOO. Other than this case, THSEE is not used but will be computed as a function of RHOO and included as an output.

• HGRND (meters)

The height of the ground at the transmitter position above MSL.

HTRANS (meters)

The height of the transmitter above.

• HSAT (meters)

The orbital altitude of the satellite above 1SL (at zenith).

• THETMX (degrees)

THETMX is the angle measured from zenith below which the laser will not transmit for a zenith pass. For offset flight paths, it should be noted that the transmitter will point below this value. For an illustration of the engagement envelope, see figure 3.1

• ICFF (meters)

For a target that does not pass direct overhead, this input specifies the amount that the target ground track is offset from the ground track of the overhead case. It is the distance, as measured from the transmitter, to a perpendicular intersection on the ground track.

• RHCO (meters) default = f(THSEL) or RHCTRE This is the turbulence coherence length as defined by H. T. Yura. [Ref. 20]. (see turbulence section for more discussion and references)

• V0 (m/s∈c)

The atmospheric wind. The direction of the wind is parallel and opposite to the direction of target motion. Note, in the present program, VO is a constant independent of altitude.

* ACELCM (*) 0.0 to 1.0

This parameter allows the user to correct for thermal blocking as if by adaptive optics. The value entered may range from 0.0 to 1.0. If 0.0 is used, complete thermal blocking compensation will occur. Conversely, if 1.0 is extered, no compensation will be applied. The variance of the phase distortion is multiplied by this correction factor before the Strehl relation is used to compute the relative intensity reduction due to thermal blooming.

$$\frac{I}{I}_{0} = \exp(-\sigma^{2} * AOBLOM)$$
 (3.4)

• AVGSFI (*) 0.0 to 1.0

AVGSFT allows the user latitude in defining the farfield spot size to other than that indicated by GUTSAVG
analysis. A value of 1.0 would result in the peak irradiance according to the program analysis. An entered
value of .5, for example, would result in a peak irradiance 50% less than the program would otherwise indicate.
AVGSFT, then, is an adjustment factor that allows the
user to account for effects not addressed in the GUTSAVG
propagation calculations.

• SIGJII (radians)

SIGJIT is the $2\sigma_p$ variance for pointing and tracking jitter.

• ADAP (*) 0.0 to 1.0

This term is a correction factor for tilt due to turbulence. It represents the residual tilt after AO compensation. ADAP may be varied from 0.0 to 1.0. A value of 1.0 would result in no tilt due to turbulence being removed while a value of 0.0 would result in total tilt due to turbulence compensation. If full AO is selected the program will set ADAP equal to 0.0.

• NFIAGA (*) 0 cr 1

NFLAGA is a selection indicator for full zonal adaptive crtics. Enter a 0 if AC is not desired, enter a 1 if AO is desired. If AC is used, XJT, BWIDTH, and NA must be specified.

• NCISC (*) 0 cr 1

ACISC is a selection indicator for isoplanatic calculations. Enter a 1 to inhibit isoplanatic calculations; enter a 0 if isoplanatic calculations are desired.

• XJI (Watts/sterad)

XJI is the target radiant intensity. It is one of the factors used to determine how noise affects the response of the AO system.

· EWIDIE (Hz)

EWILIH is the bandwidth of the adaptive optics system.

• NA (*)

NA is the number of AO actuaters used to perform phase adjustment.

· AESICZ (*)

This is the percent transmission at zenith at the sensing wavelength of the AO system. This term is used in the determination of relative noise at the AO sensor.

• N1 (*)

The number of iteration steps for the angle lcop of the program from THEIMX to 0.

• N2 (*)

The number of altitude intervals for absorption and scattering determination.

(*) EK •

The runter of altitude intervals for the turbulence calculations.

• N4 (*)

The number of iteration intervals for the MTF calculations.

• N5 (*)

The number of intervals for the slant path update of thermal blooming.

E. FNGAGEMENT GEOMETRY

The target engagement geometry is that of a earth-based transmitter and a target satellite at a given orbit. No attempt is made to define an engagement envelope based on threshold irradiance or fluence. The input parameters defining the engagement window in GUTSAVG are THETMX, ICFF, and HSAI. The general geometry of this window is shown in Figure 3.1. Only half of the total transit window is addressed in the program calculations; the results are the

¹ The GUTSFP (fcctprint) version of guts was writter to dc this. Except fcr this feature, the propagation calculations are identical to GUTSAVG.

same for either half. The program output reflects this half-window evaluation except for parameters such as total fluence and shot time which are simply double the computed values.

Figure 3.2 defines the earth center angle (ECANG). FCANG is a function of the user input THETMX. Most of the geometric calculations are referenced to earth center. Therefore, this argle is used for computin, such positioning data as the angle interval at which the irradiance will be evaluated (see Figure 3.3).

Cffset flight paths require a coordinate translation as shown in Figure 3.4. Position and velocity relative to the transmitter are computed as in Figure 3.5. It should be noted that if the flight path is offset, the zenith angle will exceed THETMX for part or all of the window. This is because the window is defined in the x-z plane only.

 H_{s} - Zenith altitude of target

 $\theta_{\rm m}$ - Maximum zenith angle for engagement (zenith path)

L - Offset distance

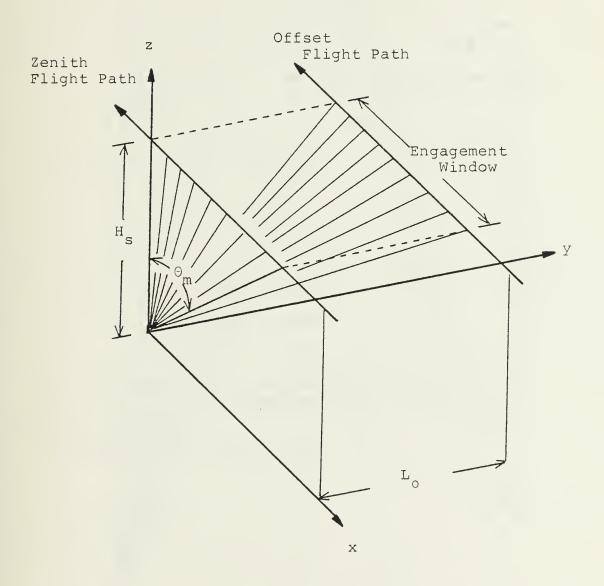


Figure 3.1 General Engagement Geometry.

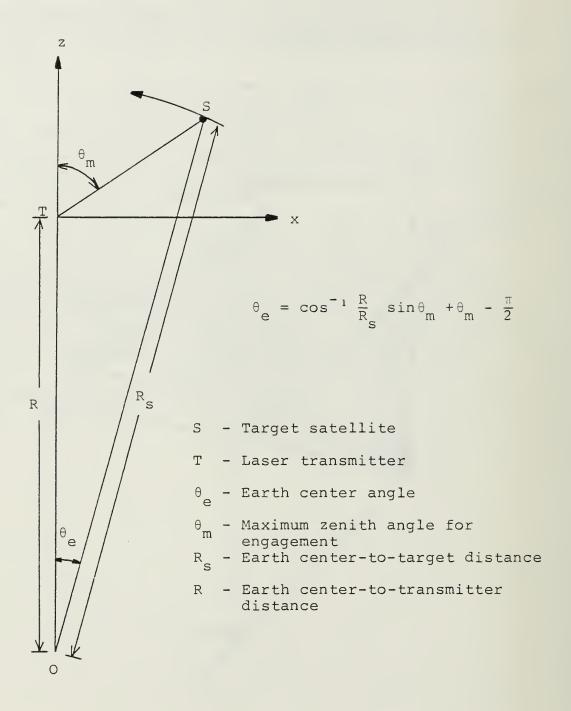


Figure 3.2 Earth Center Angle (ECANG).

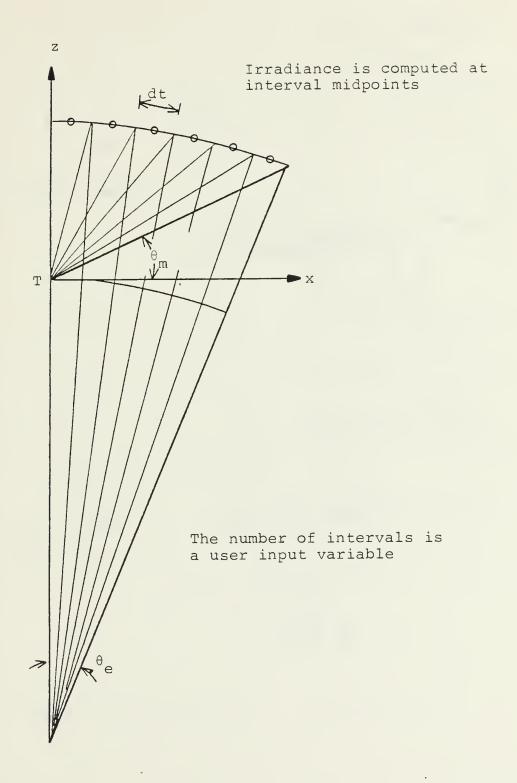


Figure 3.3 Angle Intervals for Irradiance Evaluation.

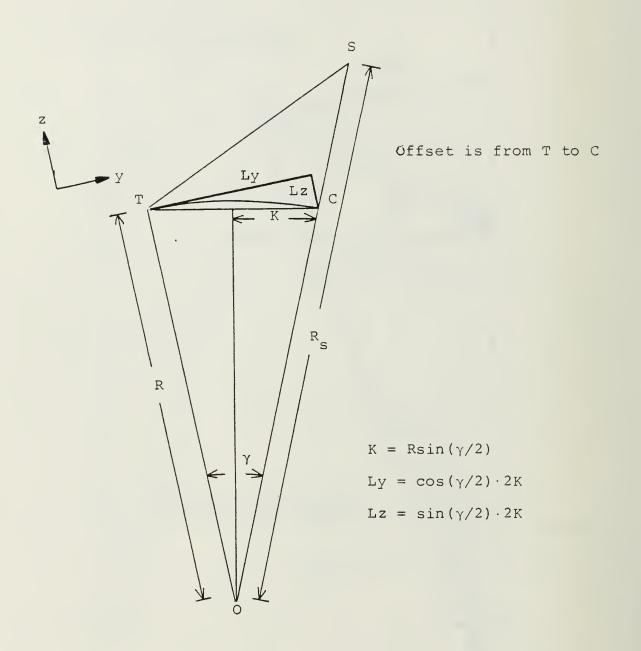


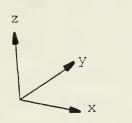
Figure 3.4 Coordinate Translation due to Flight Path Cffset.

Velocity components

$$V_{x} = -V_{s} \cos(\theta_{e})$$

$$V_{v} = V_{s} \sin(\theta_{e}) \cos(\gamma)$$

$$V_z = V_s \sin(\theta_e) \sin(\gamma)$$



T

R

Vs

Position

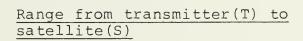
$$X = R_s \sin(\theta_e)$$

$$Y = Ly \cdot cos(\gamma) + (Z_0 - R - Lz)$$

$$\cdot$$
sin(γ)

$$Z = -Ly \cdot sin(\gamma) + (Z_0 - R - Lz)$$

where
$$Z_0 = R_s \cos(\theta_e)$$



$$R_{ts} = Z + X + Y$$

(For Ly and Lz definition, see Figure 3.4)

Figure 3.5 Velocity and Position Relative to Transmitter.

C. MAIN PECGRAM FLOW DIAGRAM

The main program mostly consists of geometry calculations and decision flow points. The decision points allow branching to adaptive optics and isoplanatic subroutines. Most of the propagation calculations are done within the subroutines. Only two major iteration loops reside in the main program, the angle interval loop and the combined MIF integration loop. The following is a general flow diagram for the main program. Decision variables are user inputs or program defined parameters.

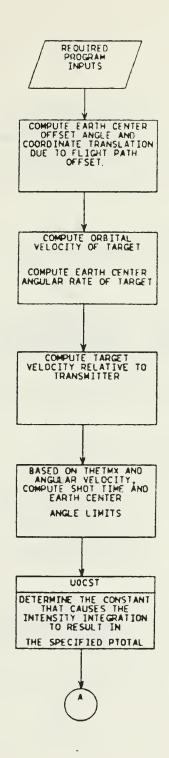


Figure 3.6 Main Program Flow Diagram.

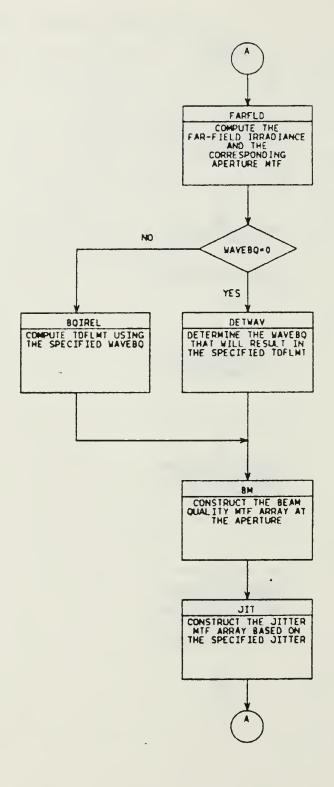


Figure 3.7 Main Program Flow Diagram (cont).

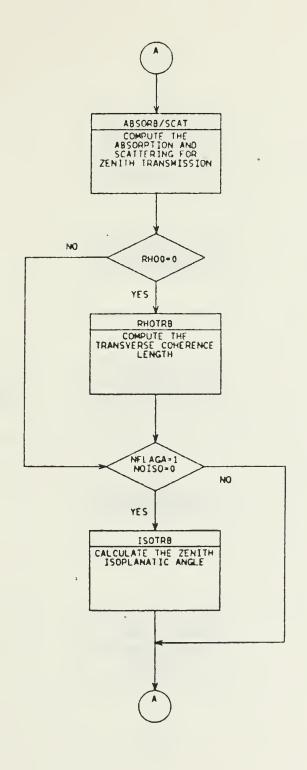


Figure 3.8 Hain Program Flow Diagram (cont).

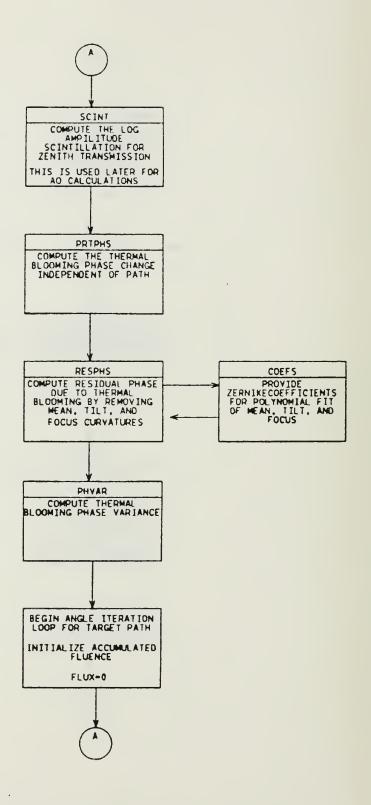


Figure 3.9 Main Program Flow Diagram (cont).

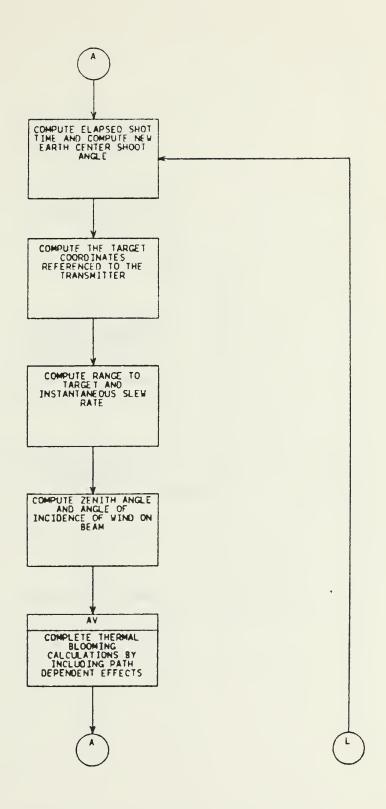


Figure 3.10 Hain Program Flow Diagram (cont):

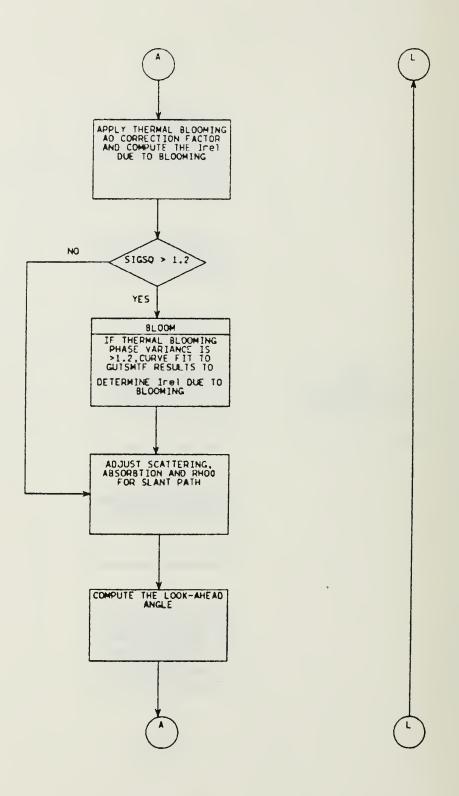


Figure 3.11 Main Program Flow Diagram (cont).

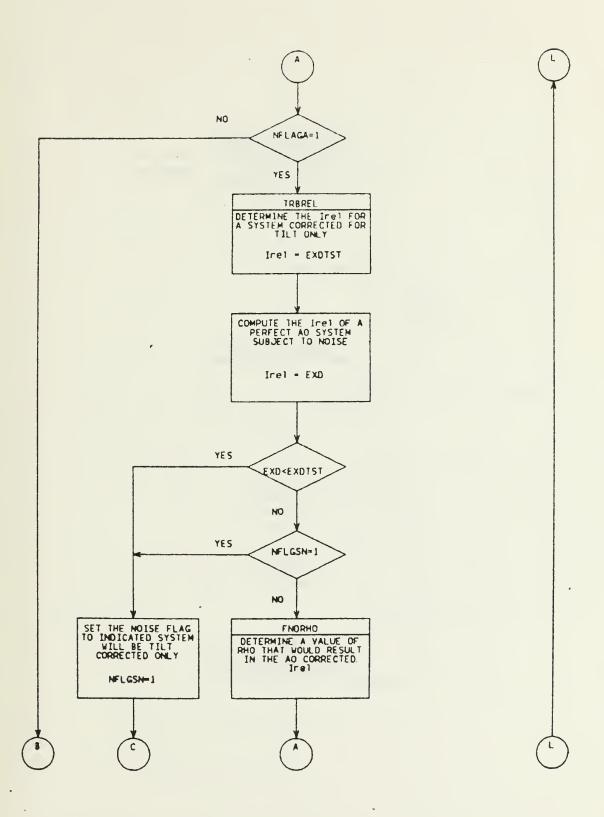


Figure 3.12 Main Program Flow Diagram (cont).

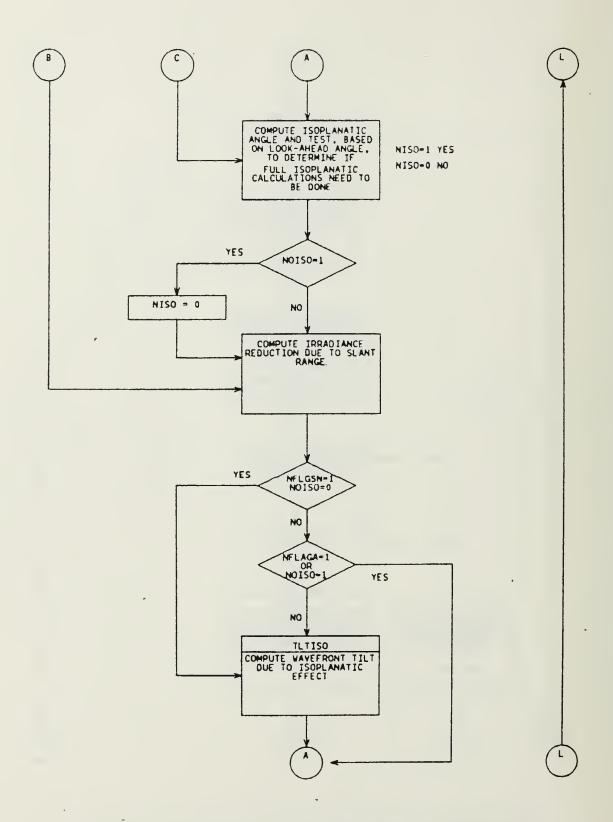
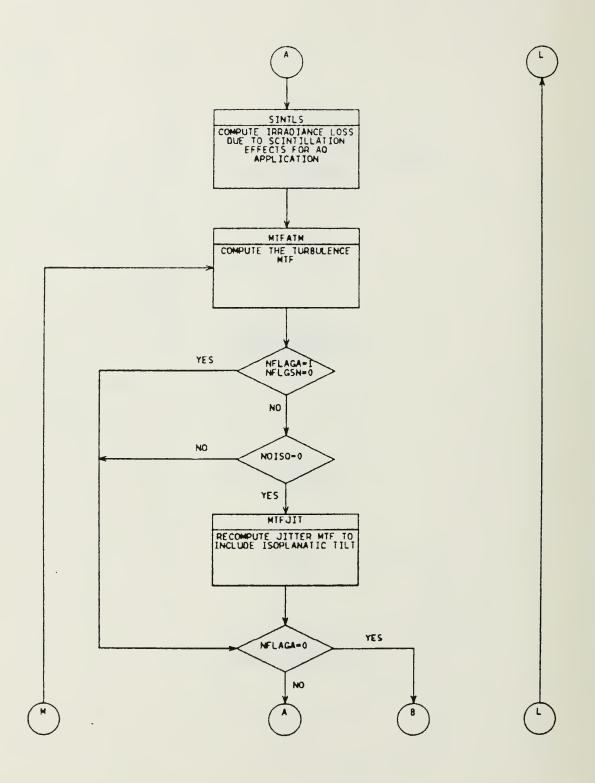


Figure 3.13 Main Program Flow Diagram (cont).





· Figure 3.14 Main Program Flow Diagram (cont).

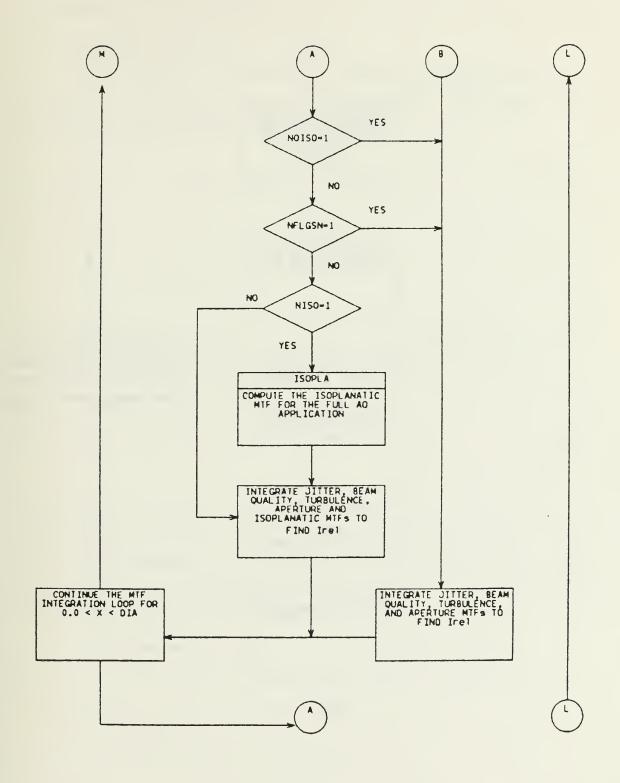


Figure 3.15 Main Program Flow Diagram (cont).

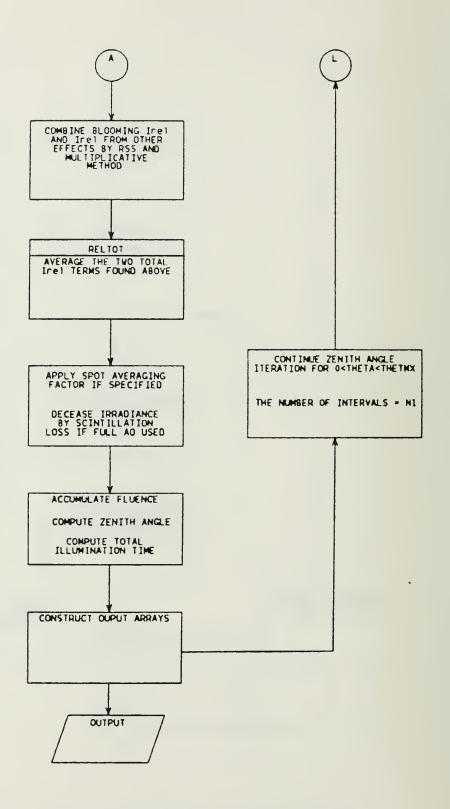


Figure 3.16 Main Program Flow Diagram (cont).

D. SUEBCUTINE DESCRIPTION

1. AESCRB and SCAT

As noted in the preceding chapter, absorption and scattering effects are treated indentically. The AESCRE and SCAT algorithms perform the integration

$$T_{a} = \exp - \int_{h_{t}}^{h_{atm}} \alpha(h) dh$$
 (3.5)

$$T_{s} = \exp -\int_{h_{t}}^{h_{atm}} \sigma(h) dh$$
 (3.0)

The result is the transmission due to the total absorption or total scattering. The correction for slant range is applied within the angle interval loop of the main program and is simply

$$T = T (3.7)$$

is the zerith angle of the target with respect to the transmitter.

The extinction coefficients for absorption (α) and for scattering (σ) are provided to the subroutines by a call to routines ALFA and ALFS respectively.

TABLE I
AESCRB and SCAT Program Variable Definitions

<u>Variabl∈</u>	<u>Fortran</u>	na re Main	<u>Definition</u>
I _a (AESCEB)	T	TABSC	Total molecular transmission
Is (SCAI)	T	TSCAIO	Total scattered transmission
h (meters)	HAIMC	HAIMC	Height of atmosphere
h _t (meters)	нт	HIRANS	Height above MSI of transmitter
α(h) (kπ) ⁻¹	ALP	-	Absorption coef. at given altitude
$\sigma(k) (k\pi)^{-1}$	ALS,	-	Scattering coef at given altitude
-	И	N 2	<pre># of integration intervals</pre>

2. AIFS and AIFA

These two routines are also identical. They provide the extinction coefficients to ABSORB and SCAT for a specified altitude. At each altitude where a coefficient value is desired, a linear interpolation is performed between data points supplied by the user. Data statements precede each of these subprograms, and it is with these statements that the absorption and scattering data should be entered. Units for the coefficient and corresponding altitude should be km⁻¹ and km respectively.

TABLE II

ALFS and ALFA Frogram Variables and Definitions

<u>Variable</u>	<u>Fortran</u>	name Maln	<u>Definition</u>
o(h) (kn) ⁻¹	S	AIS	Scattering coef. at specified altitude
$\alpha(h) (k\pi)^{-1}$	A	AIP	Absorption coef. at specified altitude
h (meters)	Н	-	specific altitude
-	ALT (NI)	-	Altitude data list
_	ATA (NI)	-	Absorption data
-	ATS (NI)	-	Scattering data
-	NL	-	Number of points in data list

3. IOCSI

For a gaussian beam with constant phase, the initial amplitude distribution at the transmitter aperture is

$$U(r) = U_0 \exp \left[-(r/w)^2\right]$$
 (3.8)

where U_0 is the amplitude and w is the spot size [Ref. 21]. The purpose of UOCST is to compute the constant U_0 for a given aperture power.

Ey the scalar wave approximation, the intersity distribution is [Ref. 22]

$$I(r) = |U(r)|^2$$
 (3.9)

To relate the field distribution to the power, the intensity distribution is integrated over the aperture.

$$P_{t} = \pi \int_{r_{i}}^{r_{0}} I(r) dr^{2}$$
 (3.10)

Substituting 3.8 and 3.9 into 3.10

$$P_{t} = \pi \int_{r_{i}}^{r_{0}} U_{0} \exp \left[-(r/w)^{2}\right] dr^{2}$$
 (3.11)

then rearranging terms, produces the expression for U_0 .

$$U = \left[P_{t} \left(\pi \int_{r_{i}}^{r_{0}} \left(\exp\left[-(r/w)^{2}\right]\right)^{2} dr^{2}\right)^{-1}\right]^{\frac{1}{2}}$$
(3. 12)

The integration limits r_0 and r_1 are the radius of the transmitter and the chscuration, respectively.

UOCST performs the integration in 3.12 using the trapezoidal rule. A call is made to subroutine FIELD to evaluate $\exp\left[-\left(r/w\right)^2\right]$ which is simply U(r) with U₀ = 1. For this reason, U₀ is defined as unity when FIELD is called by UOCST.

TAFIE III
UCCST Program Variables and Definitions

Variatle U o (%) 1/2 m - 1	Fortian Sub	na re Main -	Definition Normalization constant
F _t (Watts)	P	PICTAL	Total aperture exit power
-	N	MI	# of integration increments
r _o (reters)	OE	_	Outer radius of transmitter
r _i (meters)	RI	-	Radius of opscuration
U(I) (W) m 1	UR	-	Field amplitude at radius r
w (meters)	BMRAI	-	Radius of spot at aperture

4. FIFID

FIEID calculates the field distribution for a axisymmetric Gaussian beam with constant phase.

$$U(r) = U_0 \exp \left[-(r/w)^2\right]$$
 (3.13)

TABLE IV
FIELD Program Variable Definitions

<u>Variatle</u>	Fortran Sub	<u>na me</u> <u>Main</u>	<u>Definition</u>
U(r) (W) 1/2 m ⁻¹	UR	-	Field amplitude at r
$U_0 (K)^{\frac{1}{2}} \pi^{-1}$	U 0	-	Normalization constant
w (meters)	BMRAI	-	Radius of spot at aperture

5. FARFLD

In the Fraunhcfer region, the amplitude distribution can be found by taking the Fourier transform of the aperture distribution [Ref. 23]. Using polar coordinates and noting axial symmetry, $U(\rho)$ in the far-field can be expressed as

$$U(\rho) = \frac{1}{\lambda z} \int_{a}^{b} \int_{0}^{2\pi} U(r) \exp \left[-\left(\frac{i 2\pi}{\lambda z} \right) \cdot r \rho \cdot \cos \theta \right] r dr d\theta$$
 (3.14)

Using integral relation for the zero order Bessel function, equation 3.14 can be simplified to [Ref. 24]

$$U(\rho) = \frac{2\pi}{\lambda z} \int_{a}^{b} U(r) J_{0}\left(\frac{2\pi\rho r}{\lambda z}\right) rdr \qquad (3.15)$$

The irresity in the far-field is then given by I $(r) = |U(r)|^2$ or in terms of equation 3.15

$$I(\rho) = \left(\frac{2\pi}{\lambda z}\right)^2 \left[\int_a^b U(r) J_0\left(\frac{2\pi\rho r}{\lambda z}\right) r dr \right]^2$$
 (3.16)

where 1 and a are the radius of the transmitter and the radius of the obscuration, respectively. [Ref. 25]

FARFID evaluates 3.16 at a specified number of increments in the far-field and assigns these values to an array F(i). F(i) is remalized with the total aperture power so that the MTF produced would be unity at the origin. Note, however, that when the MTF is computed in this case, the MTF is normalized so as to produce an Irel (intensity relative) value when integrated.

$$M_{a}(\overline{\rho}) = \frac{2\pi}{(\lambda z)^{2}} \frac{P_{t}}{I_{0}} \int_{0}^{r} I(\overline{\rho}) J_{0}\left(\frac{2\pi\overline{\rho}r}{\lambda z}\right) \overline{\rho} d\overline{\rho}$$
 (3.17)

I is determined by evaluating equation 3.16 with r=0.0.

TABLE V
FARFLD Program Variables and Definitions

<u>Variabl∈</u> k(m) ⁻¹	Fortian Sub	<u>na m∈</u> <u>Main</u> -	<pre>Definition (2π/λ)</pre>
ρ	R 1	-	Radius in far-field
r	R O	-	Radius in aperture
I _o (Watts)	PMAXC	PMAXO	Far-field on-axis intensity for a diffraction limited beam
I(r) (W/m²)	F(I)	TISC	Irradiance in far-field
M _a (ρ)	G(I)	IRRMIF	Aperture MTF
-	DX2	DΧ	Aperture MTF increments
F _t (Watts)	P	PICTAL	Total power in the aperture

6. ECIFEL

If the 'times diffraction limited number' (TDFLMI) is not specified, this routine computes the Irel value due to team guality which is then used to compute TDFLMT in the main program. The calculation performed is

Irel =
$$2\pi \int M_{a}(\overline{\rho}) M_{b}(\overline{\rho}) \overline{\rho} d\overline{\rho}$$
 (3.18)

where M_a is the afertire MTF and M_b is the MTF for the beam quality phase screen. M_a is defined in subroutine FARFID so that this integration produces an Irel value. M_b is the beam

quality MIF as computed by subroutine MIFBQ. IDFLMI is computed in the main program and is 1/ Irel.

TABLE VI ECIREL Program Variables and Definitions Fortian name
Sub Mai Variable Main Definition Times diff. limited # N TDFLMI Irel REL TEQ $M_{a}(\overline{\rho})$ A(I) IRRMIF Aperture MTF Mh (P) F(I)MTF for phase screen # of iterations
for MTF calculations N4 N 4

7. LEIKAV

If the RMS phase distortion parameter (WAVEEQ) is not specified, this routine computes it based on the approximation

$$Irel = exp (-\sigma^2)$$
 (3.19)

where $\sigma=\frac{2\pi\delta_{rms}}{\lambda}$ and δ_{rms} is the RMS value of the phase distortion at the aperture. Irel is the intensity degradation due to beam quality and is equal to 1/(N)². N is the input parameter IDFLMI. DETWAV first evaluates σ^2 using equation 3.19 and ther uses this value as a starting point

to compute a more accurate σ^2 by iteration using subroutine EQIRFI. σ^2 is used to compute the beam quality MTF and the leam quality Irel as in equation 3.18. σ^2 is adjusted, and the process repeated until the Irel found by equation 3.18 is equal to that determined by $1/(N)^2$.

DETWAV	Program	TAFLE VII Variables a	and Definitions	
$\frac{\text{Variable}}{\left[\frac{2\pi\delta_{\text{rms}}}{\lambda}\right]^2}$ $\frac{\delta_{\text{rms}}}{\delta_{\text{rms}}}$	Fortran Sub VAREÇ WAVEEQ	Name Main VAREQ WAVEEQ	<u>Definition</u> Constant Phase distortion	
λ 1/N²	RELO	-	parameter 1/(IDFLMI) ²	

8. MIFEO and BM

These two routines calculate the beam quality MTF array. The MTF is [Ref. 26]

$$M_{b}(\overline{\rho}) = \exp\left(-k^{2}\left[\sigma^{2} - C_{\phi}(\overline{\rho})\right]\right)$$
 (3.20)

where $C_{\varphi}(\rho)$ is the autocorrelation function of the phase and is the phase variance. $C_{\varphi}(\rho)$ is assumed to be Gaussian. Letting $C_{\varphi}(\rho) = \sigma^2 \exp\left[-(\rho/L)^2\right]$, where L is the phase correlation length, results in [Ref. 27]

$$M_{b}(\overline{\rho}) = \exp\left[-\left(\frac{2\pi\delta_{rms}}{\lambda}\right)^{2} \left(1-\exp\left[-\left(\overline{\rho}/L\right)^{2}\right]\right)\right]$$
 (3.21)

 $\frac{\delta_{\rm rms}}{\lambda}$ is the wavelength RMS phase distortion and is ar input parameter (WAVFEQ). I ,also a user input (SCAIEQ), will default to 1/5 the diameter of the aperture if not otherwise specified.

	ŗ	TABLE VIII	
MIFEÇ and	BM Frogra	am Variables	s and Definitions
$\frac{\text{Variable}}{\left[\frac{2\pi\delta_{\text{rms}}}{\lambda}\right]^2}$	Fortran is	na me Maln V ARBÇ	Definition Beam quality variance
رم) م	F (I)	BÇMIF	Beam quality
-	N 4	м 4	# of MTF integration increments
k (m) ⁻¹	-	-	(2π/λ)

9. <u>ISCIRB</u>

ISCIRB calculates the zenith isoplanatic angle for use in the adaptive critics portion of the program. The angle is given by [Ref. 28]

$$\Theta_0 = .314 \left[\frac{2.91}{6.88} \sec \theta \int_{h_t}^{h_{atm}} C_n^2(h) h^{5/3} dh \right]^{-3/5}$$
 (3.22)

For the rear zenith case, $\sec\theta=1$. C_n^2 is the refractive index structure constant and is calculated by subroutine CN2H as function of altitude. See Fried [Ref. 29] for a discussion of the above angle.

		TABLE IX		
ISOTR	B Program	Variables	and Definitions	
<u>Variacles</u>	Fortian Sub	Name Main	<u>Definition</u>	
⊖ ₀ (rad)	ISOANG	ISCANO	Zenith isoplanatic angle	
C2 (I)	CN2	-	Refractive index structure constant	
-	N	Еи	Integration intervals for turbulence	٥
θ (đ € ς)	-	OMEGA	Zenith angle	

10. SCINT and SINILS

If full adaptive optics are used, amplitude scintillation effects will act to degrade the performance of the

conventional phase correcting adaptive optics system. This subroutine computes the log amplitude variance of the scintillations for zenith transmission. The approximation used is

$$\sigma_z^2 = .56k \int_{h_+}^{h_a tm} \int_{h_c}^{s_6} C_n^2(h) dh$$
 (3.23)

For the cff zenith case

$$\sigma_{\mathbf{Z}}^2 = \sigma_{\mathbf{Z}}^2 \operatorname{sec}(\theta) \tag{3.24}$$

where θ is the zenith angle [Ref. 30]. The relative irradiance loss due to scirtillation is

AMP LOSS =
$$\exp\left(-\sigma_z^2 \sec(\theta)\right)$$
 (3.25)

The correction for the off zenith case is applied in subroutine SINTLS in the angle loop of the main program when full adaptive optics are specified. This loss factor has been shown to be limited to approximately .55, therefore, a default value is used in the main program in the case where equation 3.25 yields a value less than 0.5 [Ref. 31].

TABLE X SCINILS and SCINI Program variables and Definitions

<u>Variahle</u>	<u>Fortran</u> Sub	<u>Na ¤∈</u> <u>Mair</u>	<u>Definition</u>
k (m) ⁻¹	СК	-	2π/λ
$C_{\mathbf{z}}^{\mathbf{n}}$ (\mathbf{z})	CN2E	-	Refractive index structure constant
$\sigma_{\mathbf{X}}^{2}$	SIGXZ	SIGXZ	Log amplitude variation
Ant loss	TAME	TAMP	loss caused by scintillation

11. CN2E

CN2H computes the index of refraction structure constant as a function of altitude. C_n² is determined using Hufnagel's model with an added term to include the near surface effects. [Ref. 32] [Ref. 33]

$$C_n^2 = (5.94 \times 10^{-53}) h^{10} \exp \left(\frac{-h}{1000}\right) + (2.7 \times 10^{-16}) \exp \left(\frac{-h}{1500}\right) + (10^{-14}) \exp \left(\frac{-h}{100}\right)$$

Here, the units of h are meters.

12. FHOIRE

This subroutine computes Yura's [Ref. 34] lateral coherence length, ρ_0 , in terms of Fried's [Ref. 35] coherence diameter, r_0 .

$$\rho_0 = r_0/2.1 \tag{3.27}$$

Substituting Fried's definition for r

$$\rho_0 = \frac{1}{2.1} \left[\frac{2.91}{6.88} k^2 \int_{h_t}^{h_{atm}} {c_n(h) dh} \right]^{-3/5}$$
 (3.28)

TABLE XI RHOTRB Program Variables and Definitions Fortian Name Sub Main Variable Definition $k (E)^{-1}$ CK $2\pi/\lambda$ C2 (I) CN2 Refractive index structure constant po (TeteIS) RHC RECO Coherence length # of integration
intervals for N N3turbulence

13. III and MTFJII

JIT AND MTFJII function algorithmically the same as EM and MTFEQ. Together, they compute the jitter MTF array. MTFJII is called by JIT for each radial increment. The jitter MTF is given by

$$M_{j}(\overline{\rho}) = \exp\left(\frac{k^{2}\overline{\rho}^{2}(2\sigma_{p})^{2}}{8}\right) \qquad (3.29)$$

where $2\sigma_p$ is a user specified input. As with beam quality, this quartity is applied at the aperture.

TABLE XII

JII and MIFJIT Frogram Variables and Definations

Variable Fortram Name Main Definition

k (m) 1 CK - 2\pi/\lambda

L (\overline{\rho}) F JIIMIF Jitter MIF

14. FRIEHS

This subroutire calculates that part of thermal blocking act dependent on the slant path of the beam. The equation used by GUTSAVG for the phase distortion due to thermal blocking is

$$\Delta \phi(x,y) = \left[\frac{2\pi}{\lambda} \frac{n-1}{p_0} \frac{\gamma-1}{\gamma} \int_{-\infty}^{x'} I(x',y) dx' \right]$$

$$\times \left[\int_{h_t}^{h_{atm}} \frac{\alpha(h) \exp\left(-\sec\theta \int_{h_t}^{h} \alpha(h) + \sigma(h) dh\right)}{V_0 \cos\xi + \omega h} dh \right] (3.30).$$

Note that the first term in this expression is invariant with respect to beam path while the second term contains such path dependent variables as wind, wind due to slew, and extinction coefficients. The first term is evaluated by this subroutine and the remainder evaluated later in the program inside the shoot angle iteration loop by subroutine AV.

Standard values used for n₀-1, p₀, and Y are 2.72x104, 1.01x105 J/m³, and 1.4, respectively. A phase screen, F(i,j), is constructed by iteration of the intensity integral. Note that program performs this integration over a half plane of the aperture defined by -b<y
b and 0<x
ch where t is the aperture diameter. The results are adjusted later in subroutire PHVAR for this method.

PE	TPHS Program	TABLE XIII	I and Definitions
<u>Variable</u> k (I) ⁻¹	Fortran Sub CK	<u>Name</u> <u>Main</u>	<u>Definition</u> 2π/λ
n ₀ - 1	const.	-	Refractive index term (2.72x104)
Υ	const.	-	Ratio of specific heats (1.4)
F ₀	const.	-	Atmos. pressure (1.01x105) N/m²
φ (x,y)	PH(i,j)	PH(i,j)	Thermal blooming phase array

15. RESFHS and CCFFFS

Chee the phase screen has been constructed by subroutine FRTPHS, subroutine FESPHS removes the mean, tilt, and fects curvature. The result is the phase aberration due to thermal blooming alone. Zernike polynomials are utilized for expressing these phase distortions with subroutine COEFFS providing the required coefficients. These polynomials are

$$Z_0(x,y) = a_0$$
 (mean) (3.31)

$$2_1 (x,y) = a_1 x$$
 (x tilt) (3.32)

$$Z_2(x,y) = a_2 y$$
 (y tilt) (3.33)

$$Z_3(x,y) = a_3(x^2+y^2) + a_4$$
 (focus) (3.34)

The expansion coefficients are given below and are computed relative to a uniform aperture weighting function W(x,y). W(x,y) = 1 inside the aperture and zero elsewhere. The integrals have been multiplied by 2 to compensate for the half plane integration of subroutine PHRPHS.

$$f_0 = 2a_0 \phi(x, y) W(x, y) dxdy$$
 (3.35)

$$f_1 = 2a_1 + \phi(x, y) W(x, y) x dx dy$$
 (3.36)

$$f_2 = 2a_2 \phi(x, y) W(x, y) y dx dy$$
 (3.37)

$$f_3 = 2$$
 $(a_3(x^2+y^2)+a_4)$ $(x,y) W(x,y) dxdy$ (3.38)

The final phase correction is given by

$$\phi(x,y) = \phi(x,y) - f_1 Z_1(x,y) - f_2 Z_2(x,y) - f_3 Z_3(x,y)$$

$$-f_0 Z_0(x,y)$$
(3.39)

For a discussion of least squares fitting of Zernike polynomials, see [Ref. 36].

16. EHVAR

The phase variance due to thermal blocming is computed by this subscutine using the residual phase screen provided by RESPES. Normalized with respect to the aperture field, the variance is given by

$$\sigma^{2} = \frac{U(x,y) \phi^{2}(x,y) + U(x,y) \phi(x,y)}{U(x,y)}$$
(3.40)

TABLE XIV PHVAR, RESPHS, and CCEFS Program Variable Definitions Fortian Name <u>Variable</u> Definition PH(i,j) $\phi(x,y)$ PH(i,j) Phase array SIGSCO SIGSÇO Phase variance Zernike coeffs. A1- A4 a o a u PHMEAN The expansion coefficients $f_0 - f_3$ PHTITX PHTITY PHFCCUS relative to a uniform weighting function

17. AV

AV evaluates the path dependent part (second term) of the thermal blooming equation 3.30. (see section PRIPES) of the thermal blooming equation 5.30. (see section PRIPES) of the and of the absorption coefficient and scattering coefficient, respectively. V_0 is the user specified wind and we is the slew rate. The denominator represents the total transverse wind component across the beam. θ is the zenith angle

and ξ is the angle of attack of V_0 with respect to the Leau. V_0 has been assumed to be opposite in direction and parallel to the target to transmitter ICS slew motion. Note also that V_0 will be applied as a constant the entire length of the Leam path. This is discussed in chapter one.

		TABLE XV	
AV P	rogram Va	ariables and	Definitions
<u>Variance</u>	Fortran Sub	$\frac{\text{Name}}{\text{Main}}$	Definition
α(h) + σ(h)	ALP	-	Total extinction coefficient
$\alpha(h)$ $(km)^{-1}$	ALS	-	Absorption coeff.
s€C (θ)	SECCEG	SECCMG	secant or zenith angle
V ₀ (V 0	V O	Wind
	PHIICT	PHILCT	Slew rate
h _{atm} (meters)	HATEC	HAIMC	Height of the atmosphere
h (meters)	ΗT	HIRANS	Height of the transmitter

18. ELCCM

The purpose of BLOOM is to provide reasonable results for thermal blooming degradation when the Irel values are below approximately 0.3. In this region, the exponential Strehl relation predicts unacceptablly severe results. Therefore, when the phase variance is greater than 1.2, ELCOM computes an Irel value based on GUTSMTF results. GUTSMTF is a full wave optics propagation code using FFTs.

Curve fit polynomials were developed using GUISMIF results for both a uniform and a truncated Gaussian aperture distriintion. The resulting polynomials are

Irel =
$$\frac{1}{-.08705 + 2.9148\sigma + .1723\sigma^2}$$
 (3.41)

Irel =
$$\frac{1}{1.2877 - 2.6491\sigma + 4.09603\sigma^2}$$
 (3.42)

If the profile under consideration is a truncated Gaussiar, i.e. the aperture diameter greater than the waist diameter of the heam, then the routine uses the truncated Gaussian polynomial. Otherwise, a combination of the two are used,

Irel_{tb} =
$$\left[1-(d/b)^{2}\right]$$
 Irel_u + $(d/b)^{2}$ Irel_{tg} (3.43)

where d is the aperture diameter and b is the waist diameter of the heam. [Ref. 37]

TABLE XVI
ELOOM Frogram Variables and Definitions

<u>Variatle</u>	Fortian Sub	n <u>Name</u> <u>Main</u>	<u>Definition</u>
σ²	S	SIGSÇ	Phase variance
Irel _{tb}	T	TELCCM	Total blooming produced Irel
Ir∈l _u	TRU	-	Curve fit Irel for uniform aperture dist.
Ireltg	TRG	-	Curve fit Irel for Gaussian aperture dist.

19. MIFAIM

MIFAIM computes the atmospheric MIF and if specified, applies a tilt due to turbulence correction. The MIF is given by

$$M_{t}(\overline{\rho}) = \exp\left[-(\overline{\rho}/D)^{5/3} \left[1-d(\overline{\rho}/D)^{1/3}\right](D/\rho_{0})^{5/3}\right]$$
(3.44)

where I is the aperture diameter, ρ_0 is the coherence diameter and d is defined as 1-ALAF. ADAP is the fractional residual tilt due to turbulence and is a user specified parameter. If ADAP=0, M will represent the fully turbulence tilt corrected MTF. If ADAP=1, M will be the uncorrected MTF and may be writter as

$$M_{t}(\overline{\rho}) = \exp\left[-(\overline{\rho}/\rho_{0})^{5/3}\right]$$
 (3.45)

ALAP may be any value between 0 and 1.

TABLE XVII
MIFATM Program Variables and Definitions

<u>Variable</u>	Fortian Sub	Name Main	<u>Definition</u>
M _t (주)	F	F 3	Atmospheric MTF
po(meters)	RHO	R HO	Coherence dia.
I (meters)	D	DIA	Aperture dia.

2C. IREFEL

TREBEL computes the approximate Irel for a tilt corrected system. This value is used for comparison with the Irel produced by a prefect adaptive optics compensated system. The purpose of this comparison is to determine if noise will degrade the AO system to an extent as to make a full AC system undesirable. If this is the case, the program will apply tilt correction only to the beam.

21. FNDSHO

Subroutine FNIRHO calculates the coherence diameter that would result in the Irel value achieved by a perfect AO system. This is done by successive calls to subroutine IRBREI and iterating ρ_0 . This new ρ_0 then becomes a factor in the AC compensation. Specifically it will be used by subrottine MTFATM to produce the atmospheric MTF of the AC corrected system.

TABLE XVIII
THERFL and FNDRHC Program Variables and Definitions

<u>Variable</u>	<u>Fortran</u> <u>Sub</u>	Name <u>Main</u>	<u>Definition</u>
M + (호)	F2	-	Atmospheric MIF
(p) ta(p)	F 1	IERMIF	Aperture MTF
Irel	EXD	EXD	Irel of perfect AO system

22. IIIISO

The call to TITISO is made when adaptive optics have not been user specified and isoplanatic calculations have not been inhibited. Also, if the signal to noise ratic at the AO sensor is such that the AO system will provide tilt correction only, TLTISC is called , again, provided isoplanatic calculations have not been inhibited.

The purpose of TITISC is to include beam wander due to isoplanatism. The 2-sigma-p tilt is computed and combined with the jitter 2-sigma-p. This combined term is then used to compute the jitter MIF.

23. FELTOT

to a multiplicative and an RSS (root sum squared) approach to combining thermal blccming and the other propagation effects.

$$Irel = \frac{Irel_{rss} + Irel_{m}}{2}$$
 (3.46)

Itel is the result of the multiplicative approach

where Irel_{tb}is the Irel due to thermal blocming and Irel_o is the Irel due to keam quality, jitter, turbulence, isoplanatism and adaptive optics effects. The RSS approach is given by

$$Irel_{rss} = \left[1 + \left(\frac{1}{Irel_{o}} - 1\right) + \left(\frac{1}{Irel_{tb}} - 1\right)\right]^{-1}$$
 (3.48)

		TABLE	XIX		
RELTOT	Prcgiam	Varial	cles	and	Definitions

-	<u>Variable</u>	<u>Fortran</u> <u>Sub</u>	<u>Na r∈</u> <u>Main</u>	Definition
	Irelt	T	LAIOPT	Total Irel
	Irel _{rss}	TR	TRSS .	Irel by RSS method
	Ir∈l _m	TM	TMULT	Irel by multiplicative method
		EXD	EXD	Irel of perfect AO system
	Irel _o	-	TIRUE	Irel due to all other effects
	Irel tb		TELCCM	Irel due to thermal blooming
i				

24. ISOPLA

This routine calculates the MTF that characterizes the isoplanatic effect on the predictive or "look-ahead" adaptive optics system. The look-ahead angle is the major input parameter to this routine. The Fortran code was written by D.L. Fried [Ref. 38]. Fried developes the isoplanatic dependency of the AO system in terms of the effective antenna gain of the laser transmitter. The MTF that Fried formulates is given by

$$M_{iso}(\overline{\rho}) = \int_{-\infty}^{\infty} 144.88 \cdot \lambda^{-2} \frac{-5}{\rho} \cdot \sec(\theta) \cdot \mu(\text{vsec}(\theta/\overline{\rho}), \phi) d\phi \qquad (3.49)$$

where

$$x = vsec(\theta/\overline{\rho})$$
 (3.50)

and

$$\mu(x,\phi) = \int C_n^2 \left\{ 1 + (xh) - \frac{1}{2} \left[1 + 2(xh)\cos\phi + (xh)^2 \right]^{-\frac{5}{6}} \right\} dh$$

$$-\frac{1}{2} \left[1 - 2(xh)\cos\phi + (xh)^2 \right]^{-\frac{5}{6}} dh$$
(3.51)

 θ is the zenith angle and ν is the target lead angle. For detailed discussion of the theory and explicit development of the Fortran code, see [Ref. 39].

25. <u>JO</u>

JO computes the zero order Bessel function based on the input argument. This routine is called by FARFLE in the calculation of the far-field irradiance.

APPENDIX A

GUTSAVG INPUT FILE

INPUT DATA FILE:

LASEE:	CO CW EDL 5P(9) TRANSITION	4.99210 EICRONS
TWOER.		4. 332 TO EICHORS
CLIMATE:	MIC-LATITUDE SUMMER, CLEAR DAY	
CASE:	1	
DIFFERENCE TO CONTROL	THIESCOFE DIAMETH CENTRAL OBSCORATION DIAMETER GAUSSIAN WAIST TEFU TELESCOPE METERS CAVITY HAVELENGTE TURE TOTAL POWER THE SOLIP RACTION LIMITED THE SOLIP RACTION LIMITED THE SOLIP RACTION LENGTH THE SOLIP RACTI	S C - 5000000 E + 01 C - 5000000 E + 02 C - 5000000 E + 02 C - 5000000 E + 01 C - 5000000 E + 01 C - 100000 E + 01 C - 5000000 E + 01 C - 10000 E + 01
N1 N2 N3 N4 N5	AFSCRFTION INTEGRATION INTERVALS TURBULENCE INTEGRATION INTERVALS	30 200 100 200
N5	MIF INTEGRATION INTERVALS IBERRAL BLOOWMING INTEGRATION INTERVA	LS 100

APPENDIX B

GUTSAVG OUTPUT FILE

LASEE: CLIMATE: CASE:	CC CW EDI 5P(9) TRANSITION MID-LATITUDE SUMMER, CLEAR DAY	4.99210 MICRONS
DIA OES EFAMSZ WAVE PIOTAL THSEE	TELESCOPE LIAMETER CENTRAL OBSCUBATION DIAMETER GAUSSIAN WAIST THRU TELESCOPE CAVITY WAVELENGTE APPERTURE ICTAL FOWER TURBULENCE SEELING ARC-SEC	.150E+01 .15CE+00 .100E+02 .499210E+05 .200E+07
HERNI TDPLMT WAVEEC SCALES HISAN HISAN THEIM	TELESCOPE LIAMETER CENTRAL OBSCOBATION DIAMETER GAUSSIAN WAIST THEU TELESCOPE APERTUBE ICTAL POWER APERTUBE ICTAL POWER IURBULIENCE SELING WETERS APERTUBE ICTAL FOWER HETERS APERTUBE ICTAL FOWER IURBULIENCE SELING WETERS APERTUBE ICTAL FOWER IURBULIENCE SELING HETERS IURBULIENCE SELING HETERS IURBULIENCE SELING HETERS IURBULIENCE SELING HETERS INMES DIFFRACTICA LIMITED PHA SE CORRELATION LENGTH TRANSHITTE HEIGHT ABOVE MSI METERS MAXIMUL ZENITE ANGLE PLIGHT PATE CFFSHT WIND VELOCITY 2-SIG MA-P JITTHE BLOCMING CCERECTION (0-1) BPOOT AVERAGINE FACTOR (0-1) USE PULL ZCNAI A-C SYSTEM INHEBIT IRANS FOE A-O SENSOB TARGET RALIANT INTENSITY A-O SYSTEM EANLWICTH NUMBEE OF A-G SYSTEM ACTUATORS NUMBEE OF ANGLE INTERVALS ABSORFTION INTERVALS HERBAL BLOCWHING INTEGRATION INTERVALS THERBAL BLOCWHING INTEGRATION INTERVALS	. 300E+03 . 120E+01 . 000E+00 . 300E+00 . 300E+03 . 100E+07
LCFF BHOO VO SIGJIT ALAF AOBLCH	PLIGHT PATE CFFSET YURA'S TURE. CCHERENCE DIA METERS WIND VELOCITY 2-SIGMA-P JITTER FRAC. RESIL. TURE. TILT BLOCKING CCERECTION (0-1) -	00 CE+00 00 CE+00 00 CE+00 10 CE+00 10 CE+01
AVGSFI NFLAGA NOISC ABSICZ XJT BWIDIB NA	USE PULL ZCHAI A-C SYSIEM - INHIBIT ISCPLANATIC CAICS - ZENITE TRANS FCE A-O SENSOB - TARGET RACIANT INTENSITY W/SIEB A-O SISTEM EANLWICTH HEETZ NUMBER OF A-O SYSIEM ACTUATORS	.75 CE+00 .250E+03 .50CE+03 .1024.
NA N1 N2 N3 N4 N5	NUMBEE OF ANGLE INTERVALS ABSORFTION INTEGRATION INTEFVALS TURBULENCE INTEGRATION INTEFVALS HTP INTEGRATION INTERVALS THERMAL BLCCWHING INTEGRATION INTERVALS	30 200 100 200 100

	PATH ANALYSI	'९ छह ्ता ग ९•					
STEP	BANGE (KM)	(MRAD/SEC)	CMEGA (LEG)	TIME (SEC)	AIMOS. TRA	ANSMISSION ABSCAP	RHO (CH)
1.	999.738	7.1582	1.08	2.63	0.99802	0.53314	51.1
2.	1000.045	7.1541	2.16	5.27	0.99802	0.53302	51.0
3.	1000-659	7.1459	3.24	7.90	0.99802	0.53278	51.0
4.	1001.579	7.1337	4.31	10.53	0.99802	0.53242	51.0
5.	1002.805	7.1175	5.39	13.16	0.99801	0.53195	50.9
6.	1004.334	7.0973	6.46	15.80	0.99801	0.53135	50.9
7.	1006-167	7.0732	7.52	18.43	0.99800	0.53064	50.8
8.	1008.301	7.0453	8.59	21.06	0.99800	0.52982	50.8
9.	10 10.735	7.0138	9.65	23.70	0.99799	0.52888	50.7
10.	1013.465	6.9786	16.70	26.33	0.59799	0.52782	50.6
11.	10 16-490	6.9400	11-74	28.96	0.99798	0.52665	50.5
12.	1019.806	6.8981	12.79	31.59	0.99797	0.52537	50.4
13.	1023-413	6.8530	13.82	34.23	0.99796	0.52398	50.2
14 -	1027.305	6.8048	14.85	36.86	0.99796	0.52248	50.1
15.	1031-481	6.7537	15.86	39.49	0.99795	0.52088	50.0
16-	1035.935	6.6999	16.87	42.13	0.99794	0.51917	49.8
17.	1040-666	6.6435	17.88	44.76	0.99792	0.51735	49.6
18.	1045.668	6.5847	18.87	47.39	0.99791	0.51544	49.5
19.	1050.939	6.5237	15.85	50-02	0.99790	0.51342	49.3
20.	1056-474	6.4606	20.83	52.66	0.99789	0.51130	49.1
21-	1062.269	6.3955	21.79	55-29	0.99787	0.50909	48.9
22.	1068.319	6.3288	22.75	57.92	0.99786	0.50679	48.7
23.	1074-621	6.2604	23.69	60.56	0.99785	0.50439	48.5
24.	1081.169	6.1907	24.63	63. 19	0.99783	0.50191	48.3
25.	1087.959	6.1196	25.55	65.82	0.99781	0.49933	48.1
26.	1094.988	6.0475	26.46	68.45	0.99780	0.49668	47.9
27.	1102.250	5.9744	27.36	71.09	0.99778	0.49393	47.7
28.	1109.740	5.9006	28.25	73.72	0.99776	0.49111	47.4
29.	1117.455	5.8260	29. 13	76.35	0.99774	0.48821	47.2

5.7510 30.00 78.99 0.99772 0.48524

30.

1125.388

47.0

TRANSMISSION ANALYSIS OUTFUT:

	TERRELISSICA	ANALISIS OO	IFCI.			
STEF	TEANSEI DUE TO SPREADING	SSICK COEFFI AMF LOSS WITH AO	CIENTS TEERMAL ELCCMING	RANGE SCALE	MAX IRRAD . (KW/CM2)	FLUENCE (KJ/CM2)
1_	0.11826	1.00000	0.01522	0.9999	0.5779E-04	C. 1522E-03
2.	0.11824	1.00000	0.01521	0.9993	0.5770E-04	C. 304 1 E-03
3.	0.11821	1.00000	0.01518	0.9981	0.5753E-04	C. 4555E-03
4.	0.11817	1.00000	0.01515	0.9963	0.5726E-04	C. 6063E-03
5.	0.11810	1.00000	C.01511	0.9938	0.5692E-04	C.7562E-03
6.	0.11802	1.00000	C.01505	0.9908	0.5649E-04	C. 9049E-03
7.	0.11793	1.00000	0.01499	0.9872	0.5597E-04	C. 1052E-02
8.	0.11782	1.0000C	C.01491	0.9830	0.5539E-04	C. 1198E-02
9.	0.11770	1.00000	C.01483	0.9783	0.5472E-04	C. 1342E-02
10-	0.11756	1.0000C	C.01473	0.9730	0.5399E-04	0.1484E-02
11_	0.11740	1.00000	C.01463	0.9672	0.5320E-04	C. 1624E-02
12.	0.11724	1.00000	C.01452	0.9610	0.5234E-04	C. 1762E-02
13.	0.11705	1.00000	C_01440	0.9542	0.5142E-04	C.1898F-02
14.	0.11686	1_00000	C-01427	0.9470	0.5046E-04	C.2030E-02
15.	0.11664	1.00000	C-01414	0.9393	0.4945E-04	C.2161E-02
16.	0.11642	1_00000	C_0140 C	0.9313	0.4839E-04	0.2288E-02
17.	0.11618	1.00000	0.01385	0.9228	0.4730E-04	C.2413E-02
18.	0.11593	1.00000	C.01369	0.9140	0.4618E-04	0.2534E-02
19.	0.11566	1.00000	C-01354	0.9049	0.4504E-04	C.2653E-02
20.	0.11538	1.00000	0.01337	0.8954	0.4387E-04	0.2768E-02
21.	0.11509	1.00000	0.01320	0.8857	0.4268E-04	C. 2881F-02
22.	0.11479	1.00000	0.01303	0.8757	0.4148E-04	0.2990E-02
23.	0.11448	1.00000,	G -01286	0.8654	0.4028E-04	C.3096E-02
24.	0.11415	1.00000	0.01268	0.8550	0.3907E-04	C.3199E-02
25.	0.11381	1_00000	0.01250	0.8443	0.37862-04	C.3298E-02
26.	0.11346	1 - 00 0 0 0	0.01232	0.8335	0.3665E-04	C. 3395 E-02
27.	0.11310	1 - 00 0 0 0	0.01213	0.8226	0-3545E-04	C.3488E-02
28.	0.11273	1.00000	0.01195	0.8115	0.3427E-04	0.3578E-02
25.	0.11235	1.00000	0.01176	0.8003	0-3309E-04	0.3666E-02
30.	0.11196	1.00000	C-01158	0.7891	0.3193E-04	C.3750E-02

```
CASE TATA AND CALCULATED FACTORS:
TRANSMITTEE CLAMETER
CAVITY WAVELENGTE
APERTURE TCTAL FOWER
TIMES DIFFERACTION LIMIT
(WIDE ANGLE SCATTERING)
TRANSMITTER ALTITUDE
SATELLITE ALTITUDE
MAX ZENITE ANGLE
PLIGET PATE CFFS ET
RHOO
                                                                        4.992100 CM
2.0000 MW
                                                              =
                                                             = 300.00 M

= 1000.00 KM

= 30.00 DEG

= 0.5105E+02 CM

= 600 M/SEC

= 300.00 M

= 0.15E+01 ARC-SEC
HHOO
WIND VELOCITY .
HEIGET OF GECUND
OPT STEING AT 5500A
BLCOMING ALAFTIVE OPTICS FACTOR = 0.1000E+01 IRBADIANCE AREA AVERAGING FACTOR = 0.1000E+01
TELESCOPE CIMENSIONS:
OUTER DIAMETER =
INNER DIAMETER =
GAUSSIAN WAIST CLAMETER =
                                                                        150.00 CM
15.00 CM
1000.00 CM
RMS WAVES DISTORTION = 0.1031E+00 PHASE CORRELIATION LENGTH = 0.3000E+02 CM
FOR TEE FULL ZONAL A-C MODEL:
ZENITH TRANS. AT A-C SEBSOR
TABGET BALLIANT INTENSITY
ADARTIVE CETICS BANLWIDTE
NUMBER OF ACTUATORS
                                                                             = 0.7500E+00
= 0.2500E+03
= 0.5000E+03
= 1024.
                                                                                                              W/STER
HERTZ
THIS EUN -
IS THE EASIC CCDE WITH CNLY SOME MEASURE OF TILT CORRECTION
AND WITHCUT AN ISOPIANATIC MODEL
NUMBER OF INTERVAL STEPS FCR:
ANGULAR INTERVAL
ABSCRPTICN INTEGRATION
RHC CALCULATION
MIF CALCULATION
THERMAL ELCCHING
                                                                                                        30
200
100
200
                                                                                  =
                                                                                  =
ZENITH LCG AMELITUDE VARIANCE = 0.4905E-02
RELATIVE IFFACIANCE REDUCTION = 0.9951E+00
ZENITH LOCK-AREAD ANGLE
                                                                         = 0.5227E+02 URAD
ISOPIANATIC JITTER (2-SIGEA-P) = 0.2004E+07
                                                                        = 5.00 URAD
= 0.1000E+01
2-SIGMA-P FEAR JITTER
TURBUIENCE JITTER REJECTION
(RESIDUAL = INITAL * ADAP)
 ======= FEOPAGATION RESULTS ======
INTEGRATED FLUX ON TARGET =
TOTAL ILLUMINATION TIME =
                                                                                                   0.007499 KJ/CM2
157.97 SEC
```

APPENDIX C

GUTSAVG PROGRAM LISTING

```
NAVAL POSTGRADUATE SCHOOL VERSION 1.2
     MODIFICATION 1.2
     LATEST CHANGE DATE 22 FEB 84
     THIS VERSION OF GUISAVG HAS BEEN MCDIFIED FOR IBM 370/360 COMPATABILITY. VARIABLE NAME LENGTHS HAVE BEEN CHANGED AND AFFROAPRIATE FORTFAN CHANGES MADE. MACHINE GENERATED BREORS MAY STILL OCCUR DUE TO UNRESOLVED CDC/IBM DIFFERENCES
     THE CLOUD MCDEL HAS BEEN REMOVED FROM THIS VERSION
REAL BOMIF (30C)
REAL JITMIF (36C)
BEAL A (10C,14)
BEAL B (15)
REAL B (15)
REAL TISC (360)
REAL LOFF, MU
REAL LOFF, MU
REAL LIREMIF (30C)
REAL LY, IZ
REAL NASCRT, NU, ISCANG, ISOANO, NA
С
           INTEGER TITLE 1 (80), TITLE 2 (80), TITLE 3 (80) INTEGER DESCR (40, 60)
C
           CCMMON /AIMO/ EAIMO
CCMMON /EC/ TIFLMI, VARBO, WAVEEC, SCALEQ
COMMON /AEBFLE/ TIA, DIAOBS, BEAESZ, UO
ML=201

JJ=0

TSTOLD=0.

F4=1

PI=2.*ARSIN(1.)

MLH=ML/2+1

BATMO=3.F4
           HATMO=3.E4
TAU=0.
XMR=1.E+6
XCM=1.E+3
XK2=1.E-3
XK2=1.E-7
RID=90./ARSIN(1.)
REARTH=6.4E6
MU=3.986F14
THETSP=2.*PI/(24.*3600.)
VSURF=BEARTH*IEFISP
C
           CALL ERRSET (208,256,-1,1,1)
\\ 5,530\\ 5,530\\ 5,530\\\
                                   (TITLE1 (I), I=1,80)
(TITLE2 (I), I=1,80)
(TITLE3 (I), I=1,80)
С
                    (5,010)
(5,010)
(5,010)
(5,010)
(6,010)
(7,010)
(7,010)
(7,010)
                                   (DESCR (1,II), II = 1,50),

(DESCR (2,II), II = 1,50),

(DESCR (3,II), II = 1,50),

(DESCR (4,II), II = 1,50),

(DESCR (5,II), II = 1,50),

(LESCR (6,II), II = 1,50),

(LESCR (7,II), II = 1,50),
           READ
READ
READ
READ
                                                                             DIACBS
EEAESZ
WAVE
FTCTAL
THSEE
            READ
            READ
 COC
            (EITHER IDFLMI OF WAVEBO MUST FE DEPINED, BUT NOT BOTH)
```

```
C-
                                                                                                                                                                                                                                                                                                  CERCE (E, II) , II = 1,50 , CR (9, II) , II = 1,50 , CR (10, II) , II = 1,50 , CR (10, II) , II = 1,50 , CR (10, III) , II = 1,50 , CR 
c
                                                                                                                                                                        TOFIED OF SCALEN A SC
                                                                                          00000000
                                                                                          IF FULL AC COBRECTION IS NOT USED, THE following PARAMETERS ARE NOT REALLY USEL. CNLY BWIDTH HAS AN EFFECT, it CAUSES the ICOK AHEAD ANGLE TO BE LARGER BY TAU*FHILOT.
                                                                                                                                                                                                                                                                                                    { LESCR (23, II), II = 1,50), LESCR (24, II), II = 1,50, , LESCR (25, II), II = 1,50, , LESCR (26, II), II = 1,50),
                                                                                                                                                                        (5,510)
(5,510)
(5,510)
(5,510)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        ABSLOZ
XJI
BWIDTH
                                                                                          READ
READ
READ
READ
  0000000000000
                                                                                          TIERATION LOCF LIMITS:
N1 NUMBER OF SUPIERIA INTERVALS FROM
N2 NUMBER OF INTEGRATION INTERVALS P
N3 NUMBER OF INTEGRATION INTERVALS P
N4 NUMBER OF INTEGRATION INTERVALS P
N5 NUMBER OF SUPINIERRALS USEL FOR S
THERMAL ELGONING
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                OM O IC THETMX
FOR ABSORPTION
FOR TURBULENCE
FOR MTF
SLANT PATH UPDATE FOR
                                                                                            READ
READ
READ
READ
READ
                                                                                                                                                                            (5, 120)
(5, 120)
(5, 120)
(5, 120)
(5, 120)
(5, 120)
                                                                                                                                                                                                                                                                                                        (LESCR (27, II), II = 1,50),

(LESCR (28, II), II = 1,50),

(LESCR (29, II), II = 1,50),

(LESCR (30, II), II = 1,50),

(LESCR (31, II), II = 1,50),
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            N1
N2
N3
N4
N5
```

```
(6, 180)
(6, 180)
(6, 185)
(6, 185)
(6, 185)
(6, 185)
                                   (DESCR (25, II), II=1,50), (DESCR (26, II), II=1,50), (DESCR (27, II), II=1,50), (DESCR (28, III), II=1,50), (DESCR (29, III), II=1,50), (DESCR (30, III), II=1,50), (DESCR (31, II), II=1,50),
                                                                            BWIDTH
NA
N1
N2
N3
N4
N5
          WRITE WRITE WRITE WRITE WRITE WRITE
DATA REALIGNMENT SECTOR
          IF (SCALEC.EQ.C.) SCALBQ=DIA/5.
IF (WAVEEC.NE.O.) VAREQ=(2.*PI*WAVEBQ)**2
SGJITO=SIGJIT
IF (NFLAGA.EQ.1) ALAP=0.
IF (NOISC.EQ.1) ISCANO=1.E10
NASCRT=SCET(NA)
LOGABS=-Alog (AESLC2)
TAU=1./BWIDTH/FI
IF (RHOO.EQ. 0. AND. THSEE.NE. 0.) BHOO= (WA VE/.552-6) **1.2*(.054/THSEE
         1)
THETMX = THETMX/5ID
HSAID=HSAI-HIRANS
HAEVGD=HIBANS-EGRND
IF (HABVGD-LI.C.) HAEVGD=O.
C
          R=REARIH+HTRANS
RS=REARIH+HSAI
00000
          COMPUTE EARTH CENTER ANGLE OFF-SET
          ANGOFF=LCFF/R
A2=ANGOFF/2.
00000
          COMPUTE COORDINATE TEANSLATIONS DUE TO CFF-SET
          LY=+2.*R*SIN(A2)*COS(A2)
LZ=2.*R*SIN(A2)**2
C
0000
          COMPUTE FARTH CENTER ANGULAR RATE
           THEIDT=SCRT (MU/BS**3)
00000
          CCMPUTE CREITAL SFEED
           VSAT=THEIDT*RS
000
           ADJUST INTENTANEOUS SLEWRATE FOR ROTATION OF EARTH, assuming both are colinear.
           THETDT = AES (THEILI-THETSF)
000
           COMPUTE VELOCITY OF SATELLITE FELATIVE TO THE TRANSMITTER SITE
C
           VS=THETDI*RS
COCOC
           COMPUTE INITIAL LIMITS ON THE TIME AND ANGLE EARTH CENTER for an or line flight path
```

```
ECANG=ARCOS(R*SIN(THETMX)/RS) + THETMX-PI/2.
TIMEX2=ECANG/TEETET
TIMTOT = 2.*TIMEX2
DTIME=TIMEX2/N1
C
         IF SATELITE ALITITUDE IS AN TEF ORDER OF THE RADIUS OF THE earth, gutsfr (footprint) WORKS BETTER BY SETTING the time step instead or the argle step
C
         IF (HSAT.GT.REARTE/2.) DTIME= 10.
00000
         COMPUTE VACUUM IBRADIANCE AND APERTUBE MTF. NOTE THAT THE POINT SPACING FERF MUST BE THE SAME AS USED IN THE MTF INTEGRATION BEICW. HENCE WE DEFINE DX NOW.
c
         DX=DIA/N4
         THE CALL TO UCCSI DETERMINES THE CONSTANT THAT MAKES THE EXIT APERTURE FOWER FICTAL.
         CALL UOCST (PICTAL, MI)
CALL PARFID (DX, TISC, IRRMTF, MI, N4, HSATD, WAVE, PTOTAL, PMAXO)
000000
         IF THE WAVEBQ IS NCT SPECIFIED AS INPUT, COMPUTE ON BASIS OF TDFLMT.
         IF (WAVEEQ.EQ.C.) CALL DETWAY (IRRMTF, DX, N4)
00000
         IF TOPINT WAS ACT SPECIFIED, COMPUTE IT .
         IF (TDFLET.NE.C.) GO TO 10
CALL BQIEEL (IFFETF, 14, DX, TBQ)
TDFLET=SCET(1-/IEC)
CCNTINUE
10
C
         DEFINE BEAM QUALITY MTF ARRAY.
         CALL BM (EQMIF, DX, N4)
         DEFINE JITTER MIF ARRAY.
         CALL JIT (JITHTF, DX, N4, SGJITC, NAVE)
000
         COMPUTE ZENITE ABSORPTION, SCATTERING, AND RHOO VALUES.
         CALL ABSCEB (N2, HIRANS, TABSO)
CALL SCAT (N2, HIBANS, TSCATO)
IF (RHOO.FQ.O.) CALL RHOTRB (N3, HTRANS, HABYGD, WAVE, RHOO)
00000
         DEFINE SEEING AT 5500A
          THSEE= (WAVE/.55E-6) **1.2* (.054/RHOO)
С
        IF (NPLAGA.GE.1.AND.NOISO.EQ.C) CALL ISCIEB (N3, HTRANS, HABVGD, WAVE 1, ISOANO)
          CAICULATE LOG AMFIITUDE SCINTILIATION FOR ZENITH ANGLE.
```

```
CALL SCINT (SIGX2,N3,HTRANS,HGFND,WAVE)
000
           CCMPUTE THE THIRMAL ELOOMING FRASE DISTORTION THAT does not depend SLANT PATH CHARACTERISTICS
C
           CALL PRTFES (MI,MIH, WAVE, PH)
CALL RESPES (MI,MIH, FH)
CALL PHVAR (MI,MLH, FTOTAL, PH, SIGSQO)
C
č
           INITIALIZE ACCUMULATED PLUENCE TO ZEBO
           FLUX=0.
C
           SEI SNR FLAG IC CFF
C
           NFIGSN=0
С
C*********** M A I N
                                                 PRCGFAM
                                                                                  I C O P *************
           LCOP ON IBRADIANCES AND ACCUMULATED FLUENCES FOR A FIXED LOFF
C
           DO 120 I=1,N1
00000000
           COMPUTE TOTAL LINGTE OF ILLUMIDATION TIME FROM (+,-)
OVER PASS ANGLE
THIS IS THE TOTAL TIME. WE USE HALP OF THIS TO DETERMINE THE
PUNCTION EVALUATION ANGLE.
           TIME=2.*((I-1)*CTIMF+DTIME/2.)
COMPUTE FARTH CENTER ANGLE AT CN LINE CCORDINATES AND TIME/2. THIS ANGLE IS TO THE MID-POINT OF THE INTEGRATION INTERVAL. THE ANGLE ECANGE, COMFUTED LATER, IS TO THE UPPER LIMIT OF THE integration limit.
          ** THE COMMENTED ECANG SHOULD PE USED WEEN IT IS DESIRED THAT THE INTEGRATION SHOULD START AT THE LOW POINT AND INCREASE TOWARDS THE ZENITH. THE CURFENT CALCULATION STARTS AT THE ZENITH AND ECES DOWN.
           ECANG= (TIETOT-IIHE) /2. *THETDT
            ECANG=IIEE/2. * TEETDI
CCC
           COMPUTE TARGET COCRDINATES
C
           X0=RS*SIN (ECANG)

Z0=RS*COS (ECANG)

X=X0

Y=IY*COS (ANGOFF) + (Z0-R+LZ)*SIN (ANGOFF)

Z=-LY*SIN (ANGOFF) + (Z0-R+LZ)*CCS (ANGOFP)
 0000000
            COMPUTE ON LINE OF SIGHT ANGLE OF SAT.
THIS ANGLE IS NOT USED AT PRESENT. IT IS THE ANGLE OF THE SAT AS MEASURED FROM A POINT UNDER THE GROUND TRACK.
            THETA= ATAN (RS*SIN (ECANG) / (RS*CCS (ECANG) -R))
```

```
000
           COMPUTE BANGE IC TAEGET
CC
           BANGE = SQ ST (X * * 2 + Y * * 2 + Z * * 2)
00000
           INSTANTANEOUS SIEW BATE
           VX=VS*COS (ECANG)
VY=-VS*SIN (ECANG) *SIN (ANGOFF)
VZ=-VS*SIN (ECANG) *CCS (ANGOFF)
WX=-VYZ-2*VY
C
           PHIDOT = SCRT (WX ** 2 + WY ** 2 + WZ ** 2) / RANGE ** 2
000000000
           COMPUTE ANGLE FICE ZENITH
           OMEGA EQUALS ANGLE DOWN FROM ZENITH OMWIND EQUALS ANGLE OF WIND ATTACK TO LCS IF TARGET MOTION AND WIND ARE COPLANAR.
           OMEGA = ARCCS(Z/FANCE)
CMWIND = ARSIN (X/RANGE)
С
           COSOMG=CCS(OMEGA)
SECOMG=1./COSCMG
COSWND=CCS(OMWIND)
00000000
           ADJUST FOR SLANT FATH, BLOOMING LOSS, AESCRPTION LOSS, SCATTERING LOSS, AND TUREULENCE RHO
           CALL AV (N5, VC, FHIDCT, HTRANS, CCSWND, cosong, E) SIGSQ=SIGSQ0*F
000000
           APPLY THERMAL ELCCHING ADAPTIVE OPTICS DEGREE OF COMPENSATION
            SIGSQ=SIGSQ*ACEICE
C
           SIG=SQRT(SIGSC)
TELOOM=EXF(-SIGSC)
IF (SIGSC.GT.1.2) CAIL BLOOM (FIA, BEAMSZ, SIGSQ, TBLOOM)
TABS=TABSO**SFCCMG
TSCAT=TSCATO**SECCMG
BHC=RHOO*COSCMG**(.6)
000000
   COMPUTE LOOK AHEAD ANGLE ASSUMING FARTH ROTATION AND SATELLITE MOTION ARE IN THE SAME DIFECTION.
           TWCT=RANGE*2./3.EE
DRI=TWOT* (ABS (VSAI-VSURF))
PRCJAN=AECOS (X/EANGE)
PECJ=DRI*SIN (FECJAN)
NU=PROJ/EANGE+TAU*PHIDOT
IF (I.EQ.1) ZENNU=NU
000
            IF executed, the following is a full ao simulation
            IF (NPLAGA.EQ.C) GC TC 40
```

```
MCDIFY REC FOR EFFECT OF ADAPTIVE OPTICS
        RHCLD=RHC
RHOU=RHO
RO=2.1*RHO
000000
        DETERMINE APPROXIMATE I-REL TO SEE IF AC SHOULD BE USED, given PRESENT NOISE.
        DCE=DIA/EHO CALL TRBEEL (IFRMIF, LIA, N4, DX, EXDTST, RHC)
000000
        COMPUTE RESIDUAL VARIANCE DUE TO PERFECT ADAPTIVE OPTICS, INFINITE FANDWILLE, AND FINITE NUMBER OF ACTUATORS.
        D1=_320* (DIA/RO/NASQET) ** (1_6667)
TES=LOGAES*SZCCMG
TSENSR=PNF(-TES)
        DETERMINE PHASE VARIANCE ASSOCIATED WITH SENSOR.
CC
       PHSERR = 8.5E-6*EWIDTH*NA**2*(RANGE/5.E5*2.5/DIA)**4*(80./XJT*.5/TSE1NSE)**2
        INCREASE RESIDUAL PHASE VARIANCE AND COMPUTE THE I-REL.
        D1=D1+PHSERE
        COMPUTE I-REL CH AC TURBULENCH CORRECTED BEAM.
        EXD=EXP (-D1)
000000
        IF AO CORBECTED I-REL IS LESS THAN NO AC CORRECTION, ASSUME ONLY TILT CORRECTION IS USED.
        IF (EXD.11. EXDISI.OR.NFLGSN. EQ. 1) GO TO 20
С
        CALL PNDEBO (EXD, RHO, DIA, RHOU, DX, N4, IRRMTP)
C
        GC TO 30
CONTINUE
200000
         THIS SETS A FLAG TO FRINT THAT AO SYSTEM IS ONLY TILT
         RHC=RHOLD
NFLGSN=1
0300000000
         CONTINUE
         ESTABLISH TEST INCREMENTS TO SEE IF FULL ISOPLANATIC CALCULATION NEEDS TO BE DONE.
         COMPUTE THE ISCPLANATIC ANGLE.
         ISOANG=CCSOMG**(2.6667)*ISOANO
TSINEW=NC/ISOANG
TSIDIF=AES(TSINEW**1.666667-TSICLD)
```

```
EISDIP=EXF(TSICIP)
IF (NISO.EQ.1) ISTOLE=TSTNEW**1.666667
000
              DETERMINE IF ISCHLANATIC MTF SECULD EE BE-COMPUTED
CC
              NISO=0
IF (ETSDIF.GI.1.1) NISO=1
IF (I.FQ.1) NISO=1
if (noisc.eq.1) niso=0
CONTINUE
40
CCC
              REDUCE PEAK IRRADIANCE BY SLANT DISTANCE
CC
              IMAXSC = (ESATE/BANGE) **2
0
         THIS ALLOWS ONE IC INCLUDE TILT ISOPLANATISM. NOTE HOWEVER, THAT. IF THE PULL ADAPTIVE CPTICS ALGCRITHM IS USED WITH THE ISOPLANATISM CORRECTION, THEN THIS CALL SHOULD NOT BE USED-REPEAT-SHOULD NOT BE USED. ALSO NOTE TEAT A VERTICAL TURBULENCE PROFILE IS NEEDED WHEN EVER THIS FOUTINE IS USED.
              IF (NPLGSN.EQ. 1. AND. NOISO.EQ. 0) GO TC 50 IF (NOISC.EQ. 1. CE.NFLAGA.EQ. 1) GO TO 60 CONTINUE
50
              OHX=NU
CHY=0.
CALL TITISO (DIA,CMX,OMY, BESTIT, SECOMG, HTRANS, HABVGD)
C
              SIGJIT=SCRT (SGJITC**2+RESILT)
CONTINUE
6000000
              CCMPUTE IRRADIANCE LECREASE DUE TO AMPLITUDE SCINTILLATION FOR THE OFF ZENITH PASS. NCTE THAT TELS AS LOSS WILL ONLY BE APPLIED IF FULL ADAPTIVE CPTICS COMPENSATION IS ASSUMED.
С
              CALL SINTLS (TAME, SIGNZ, SECONG)
00000
              CALCULATE EFFECIS OF JITTER AND TURBULENCE
              SUM=0.
DX2=DX/2.
000000
              IT IS IMPORTANTE TO NOTE THAT ALL THE ABRAYS ARE DEFINED TO BEGIN AT DY/2.
            I=DX2
DO 110 J=1,N4
IF (X.GI.DIA) GO TC 100
F1=IRRMTF(J)
CALL MTFATM (X,CIA,RHO,ADAP,F2)
F3=JITMIF(J)
IF (NFLAGA.EQ.1.AND.NFLGSN.EO.0) GO TO 70
IF (NOISC.EQ.0) CALL MTFJIT (X,SIGJIT,WAVE,F3)
CONTINUE
F4=1
IF (NPLAGA.EQ.0.CE.NCISO.EQ.1.CR.NFLGSN.EQ.1) GO TO 90
F4=TISO(J)
IF (NISO.EQ.1) CALL ISOPLA (JJ,NU,X,SECOMG,WAVE,F4,HTRANS,HABVGD,T
1ISC,N4,J)
CONTINUE
F5=BOMTF(J)
SUM=SUM+F1*F2*F3*F4*F5*X
CONTINUE
X=X+DX
70
 90
 100
               X = X + DX
```

```
CCNTINUE
TTURB=SUM*2.*FI*DX
110
0000000
            CALCULATE THE RESULTING INSTENTANEOUS INTENSITY
            CONVERT IC AN APPROCE WHICH AVERAGES THE RESULT OF AN RSS IREATMENT OF THEREAL BLOOMING WITH A MULTIPLICATIVE TREATMENT
            SSCT=1./TIURB-1.
SSCB=1./TELOCM-1.
SSC=SSCT+SSOB
TMULT=TTUBB*TELCC2
TRSS=1./(1.+SSC)
CALL RELTCT (TICTAL, TRSS, TMULT)
PMAX=PMAXO*TABS*TSCAT*TMAXSC*TICTAL
000000
            APPLY AVERAGING FACTOR APPROPRIATE TO SPCI SIZE LESIBED
            PHAX=PMAX*AVGSFT
C
            IF FULL AC IS USED, DECREASE IBRADIANCE FOR AMPLITUDE SCINTILLATION EFFECTS
            TAMPL= 1.0

IF (NFLAGA.EQ.1.AND.NFLGSN.EQ.0) TAMFL=TAMP

IF (TAMPL.LT.0.5) TAMPL=0.5

PMAX=PMAX*TAMFL
0000
             ACCUMULATED FULX
            PLUX=PLUX+PMAX*CTIME
000000
            COMPUTE INTEGRATED TIMES AND ACCUMULATED ANGLES. THE IRRADIANCE FUNCTION HAS BEEN EVALUATED AT INTERVAL MID-POINTS.
             ECANGF = (TIME+FTIME) / 2.* THETDT
XOF=RS*SIN(ECANGF)
ZOF=RS*CCS(ECANGF)
XF=XOF
             YF=LY*COS (ANGOFF) + (ZOF-R+IZ) * SIN (ANGCFF)
ZF=-IY*SIR (ANGCFF) + (ZOF-R+LZ) * CCS (ANGOFF)
RANGEF=SQRT (XF**2+YF**2+ZF**2)
C
             OMEGAP = AECOS (ZF/RANGEF)
0000000
            TIME OF ILLUMINATION FROM ZENITH TO TEETA. THIS IS ALSO EVALUATED AT THE LOWER LIMIT AND NOT THE MID-POINT. IF THE COMPUTATIONS ARE DONE FROM THE THETMX TO ZENITH INSTEAD OF THE WAY THEY ARE, THE TIME DEFINITION NEEDS TO BE CHANGED.
č
             TILLUM=TIME/2.+DTIME/2.
       A (I, 1) = I

A (I, 2) = R ANGE* XKW

A (I, 4) = PRIDOT/ XKW

A (I, 4) = OMEGAF* 6TD

A (I, 5) = TILLUM

A (I, 6) = TAES

A (I, 8) = REC*XCM
```

```
A (I,9) = TIURB

A (I,10) = IAMPL

A (I,11) = IELOCE

A (I,12) = IHAXSC

A (I,13) = FEAX * XK2

A (I,14) = FLUX * XK2

CONTINUE
120
C
C===
                       (1) = DIA * XCM

(2) = WAV F * XM W

(3) = PTOFIAL/X M W

4) = TDFIMT

(4) = HTRANS

(6) = HTRANS W

(7) = HTRANS W

(8) = HTRANS W

(9) = RHOOO*XCM

(10) = RHOOO*XCM

(11) = HGGEND

(12) = THSEE
                 C
                                 (6,220)
(6,240)
(6,230)
(6,250)
(6,330)
(6,130)
                WRITE WRITE WRITE WRITE WRITE
                                                        ((A(I,J),J=1,8),I=1,N1)
                                                       (A (I, 1), (A (I, J), J=9,14), I=1, N1)
(E (II), LL=1,12)
ACEICH, AVGS PT
C
                DIA-DIA*XCM
DIAOBS-DIAOBS*XCM
EFAMSZ-BFAMSZ*XCM
WRITE (6,140) DIA,DIAOBS,FEAMSZ
                SCALBQ = SCALBQ * YCM

WRITE (6,150) WAVFEQ, SCALEQ

E(1) = ABSICZ

E(2) = XJT

B(3) = BWILTH

B(4) = NA

WRITE (6,380) (E(II), LL = 1,4)

WRITE (6,260)

IF (NPLAGA.EQ.1.AND.NOISO.EQ.1)

IF (NPLAGA.EQ.1.AND.NOISO.EQ.1)

IF (NPLAGA.EQ.1.AND.NOISO.EQ.1)

IF (NPLAGA.EQ.1.AND.NOISO.EQ.0)

WRITE (6,340) WI, N2, N3, N4, N5
С
                                                                                                        WRITE
WRITE
WRITE
WRITE
 С
                 ESIGXZ=EXF(-SIGXZ)
WRITE (6,160) SIGXZ, ESIGXZ
 C
                 ZENNU=ZENNU*XMW
WRITE (6,320) ZENNU
 C
                 RESTLT=SCRT(RESTLI) * XMW
WRITE (6,360) RESTLI
 C
                 SGJITO=SGJITO*XMW
WRITE (6,350) SGJITC,ADAP
 C
                 PIUX=PLUX*XK2*2.
TILLUM=TILLUM*2.
WRITE (6,370) FLUX,TILLUM
IF (NPLGSN.EQ.1) WRITE (6,170)
STOP
FORMAT (10%, BLCCKING ADAPTIVE CPTICS FACTOR = ', E10.4,/,
```

```
140
150
160
170
180
185
190
195
220
230
240
250
260
270
280
290
300
310
320
330
340
350
360
370
380
```

```
C--- CAICULATE THE TOTAL SCATTERING

DIMENSION ALT (20), ATS (20)

C***** THIS CATA

UPDATE OF TEE MCCLATCHY LINE DATA. THE WAVELENGTH FOR THIS

C***** TRANSITION IS 4.99210 MICKONS.

C***** ATMOSPHERE IS MILLATITUDE SUMMER, CLEAR DAY.

DATA ALT/0.0, 1.0, 2.0, 3.0, 4.0, 5.0, 7.0, 8.0, 9.0, 10.0, 12.0, 14.0,

1 DATA ATS/1.41=0.2, 9.36E=0.4, 1.01=0.5, 1.01=0.5, 4.54E=0.5, 3.50E=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.01=0.5, 1.
```

```
CC---- CALCULATES ZENITH THE TOTAL INTEGRATED MOLECULAR TRANSMISSION

COMMON /ATMO/HATEC
DELH=(HATMO-HI)/N
HEIGHT=HI+DELH/2.
ALPZ=0.
DC 10 I=1,N
CALL ALFA (ALF, EEIGET)
ALPZ=ALPZ+ALP
HEIGHT=HEIGHT+DELE
CONTINUE
ALFZ=ALPZ*DELE
T=EXP(-ALFZ)
EETURN
END
```

```
SUBROUTINE SCAT (A, HI,T)

C---- CALCULATES ZENITH THE TOTAL INTEGRATED SCATTERED TRANSMISSION

COMMON /AIMO/ BAIRC

DELH= (HAIMO-HI)/N

HEIGHT=HI+DELH/2.

AIPS=0.

DO 10 I=1,N

CALL ALFS (ALS, FEIGHT)

AIFS=ALPS+ALS

HEIGHT=HIGHT+LELE

CONTINUE

ALFS=ALPS*DELF

1=EXP(-ALFS)

RETURN

END
```

```
CC----
COMPUTES YURAS ZENITE ATMOSPHERIC COBERENCE DIAMETER

CCMMON /ATMO/ HATMO
TPI=4.*AESIN(1.)
CK=TPI/W
CKSO=CK*CK
DELB=(HAIMO-HT)/FICAT(N)
HEIGHT=HG+DELB/Z.
DC 10 I=1,N
CALL CN2F (HEIGHT,CN2)
SUM=SUM+CN2
HEIGHT=HEIGHT+DELE
CONTINUE
CNI=SUM*CELH
DEN=2.91*CKSO*CNL
RHO=(6.88/DEN)**(.6)
RETURN
END
```

```
SUBROUTINE FARFIC (CX2, F, G, MI, N4, HSAID, WAVE, PTOTAL, PMAXO)
C---- COMPUTES THE FARFIELD IRRADIANCE OF THE APERTURE DISTRIBUTION C--- AND THEN COMPUTES THE CORRESPONDING APERTURE MED
                                     AND THEN COMPUTES THE CORRESPONDING APERTURE MEDITALISM THE CORRESPONDING APERTURE MEDITALISM THE CORRESPONDING APERTURE MEDITALISM TO THE PROPERTURE MEDITALISM THE PROPERTURE MEDITALISM THE PACTOR SOLVE TO THE PROPERTURE MEDITALISM THE PACTOR PROPERTURE MEDITALISM TO THE PROP
20
C----
                                       COMPUTE FEAXO ON AXIS
                                        SUM=0.

BO=DIAOBS/2.+LXO/2.

DC 30 I=1,ML

CALL FIELE (RO,UR)

SUM=SUM+UR*RO

RO=RO+DXC

PMAXO= (SUM*CKF*DXO)**2
 30
C C--- CNORM- FIXES THE MTF SO THAT THE MTP INTEGRATION PRODUCES AN C--- IBEL VALUE.
                                           CNORM= (WAVE*HSATD) ** 2*PMAXO/PICTAL
                  --- COMPUTE TEE MIF
                                          R2=DX2/2.

DC 50 I=1,N4

SUM=0.

R1=DX1/2.

DC 40 J=1,ML

Z=CKR*R1*F2

CALL JO (Z,A)

SUM=SUM+A*F (J) *F1

R1=R1+DX1

G(I)=SUM*IWOPI*LX1/CNORM

R2=R2+DX2

RETURN
  40
  50
                                            RETURN
END
```

```
SUPROUTINE JO (X,F)
00000
                                                         COMPUTES THE ZEFO ORDER BESSEL FUNCTION.
                                             DIMENSION A1 (16), A2 (11), A3 (11), A4 (15), A5 (11), A6 (11)

REAL*8 A1 (16), A2 (11), A3 (11), A4 (15), A5 (11), A6 (11)

DATA A1/-1.0D-15, 4.1L-14, -1.944D-12, 7.8487D-11, -2.679257D-9, 7.6081
1636D-8, -1.7615465C8L-6, 3.2460528821D-5, -4.50626166206D-4, 4.8151800
269468D-3, -.034893765411405, .158067102332097, -.37009499387265, .2651
378613203337, -.C68723442352852, .31545594294978,
DATA A2/1.0D-15, -5.0C-15, 4.3D-14, -4.3D-13, 5.168D-12, -7.3641L-11, 1.
1630646D-9, -5.1765945D-8, 3.075184788D-0, -5.36522046813D-4, 1.598892C0

DATA A3/-1.0D-15, 4.0D-15, -3.3E-14, 3.01D-13, -3.207D-12, 4.2201D-11, -1.
17.27192D-10, 1.7572457D-8, -7.41449841D-7, 6.5385199426D-5, -.03111170
29210674/

DATA A4/1.1D-14, -5.78D-13, 2.5281D-11, -9.42421D-10, 2.949707D-8, -7.6
11758781D-7, 1.586701924D-5, -.000260444389349, .003240270182684, -.C29
31.19180116054122, 1.29671754121C53/
DATA A5/-1.0D-15, -5.0D-15, -4.7E-14, 4.7D-13, -5.705D-12, 8.8169D-11, -1.
1871891D-9, 6.177634D-8, -3.9872843D-6, .000893989833086, 2.001E0060217
2003/
DATA A6/1.0D-15, -5.0D-15, 3.6D-14, -3.26D-13, 3.515D-12/-4.6864D-11, 8.71891D-11, -2.22919D-11, -2.26555781D-8, 9.13861526D-7, -9.6277235492D-5, .093555574
2139071/
ZEBC ORDER BESSEI FUNCTION

LE (N.EO.1), GC TO.50
                                                      ZEBC ORDER BESSEI FUNCTION
N=0

IF (N.EQ.1) GC TO 50
IF (ABS(X).GT.8.0) GC TO 20
Y=.0625*X*Y-2
E=0.0
BF1=0.0
DC 10 I=1,16
EF2=BF1
B=Y*BF1-EF2+A1(I)
F=.5*(B-EF2)
RFTURN
Y=.256/(X*X)-2
AE-ABS(X)
B=0.0
DC 30 I=1,11
BF2=BF1
BF1=B
B=Y*BF1-FF2+A2(I)
F=.5*(B-EF2)
B=C.0
BF1=0.0
DO 40 I=1,11
EF2=BF1
EF2=BF1
FY=BF1-FF2+A3(I)
0=4*(B-BF2)/AE
Y=AB-.786358163357448
P=.7978845608C2865*(F*COS(Y)-C*SIN(Y))/SQRT(AB)
IF (ABS(X).GT.8.0) GC TO 70
BF1=0.0
DC 60 I=1,15
BF2=BF1
BF1=B
B=Y*BF1-EF2+A4(I)
F=.0625*(E-BF2)*X
BFTURN
Y=.0625*(E-BF2)*X
BFTURN
Y=256/(X*X)-2
AE-ABS(X)
B=0.0
DC 80 I=1,11
EF2=BF1

EF2=BF1
C
  10
  20
  30
  50
  60
  70
```

```
BP1=B
B=Y*BP1-EF2+A5(I)
P=.5*(E-F2)
B=0.0
BP1=0.0
DC 90 I=1,11
BF2=BP1
BF1=B
90 E=Y*BP1-EF2+A6(I)
Q=4*(B-BF2)/AE
B=SIGN(1.C,X)
Y=AB-2.35£19445C15235
F=.7978845608C2E65*(F*COS(Y)-C*SIN(Y))/SCRT(AB)
F=B*F
RETURN
END
```

```
SUBROUTINE FIELD (R,UB)

C ---- CCMPUTES EXIT APEFTUFE FIELD DISTRIBUTION. NOTE IT MUST BE AXI-SYMETRIC.

CCMMON / ARBPLD/ DIA, DIAOBS, BEAMSZ, UO UR=0.
D=2.*R
IF (D.GT.DIA.CF.D.LT.DIAOES) FETURN
EMBAD=PEAMSZ/2.
UR=U0*EXE(-(B/EMBAD)**2)
RETURN
END
```

```
SUBROUTINE DOCST (FT,N)

C---- COMPUTES NORMALIZATION CONSTANT GIVING FTOTAL ENERGY IN BEAM

COMMON / ARBFLL / IIA, IIAOBS, BEARSZ, UO
PI=2.*ARSIN (1.)
BCSD=RO**2
BCSQ=RO**2
RI=DIAOBS/2.
BISQ=RI**2
DX=ROSQ/N
CST=PI*DA
X=DX/2.
UO=1.
SUB=0.
DC 10 I=1,N
RGOTX=SOX.GT. FC.CF. ECOTX. IT.RI) GO TO 10

CALL FIELD (RCCTX, UR)
SUB=SUB+UE**2
X=X+DX
TEMP=SUB+UE**2
TEMP=SUB+UE**2
TEMP=SUB+UE**2
TEMP=SUB+UE**1
EFTURN
END
```

```
SUEROUTINE MITAIM (X,D,RHO,ADAF,F)

C---- CCMPUTES ATMOSPEFRIC MIT FUNCTION

XC=X/D
A1=1.-ADAF
DRHO=D/RHC
F=EXP(-XC**(1.6667)*(1.-A1*XD**(.3333))*DRHO**(1.6667))
RETURN
END
```

```
CC---- CCMPUTES JITTER MIF FUNCTION

IPI=4.*ARSIN(1.)
CK=TPI/WAVE
F=EXP(-(CK*X*SIGJIT)**2/8.)
REJURN
END
```

```
SUBROUTINE MTFEC (X,F)

C---- CCMPUTES RANDOM PHASE MTF. ASSUMES A GAUSSIAN CORRELATION

C---- FUNCTION.

CCMMON /EC/ TIFLMT, VARBO, WAVEFC, SCALBO
TEMP=1.-FXP(-(1/SCALEQ)**2)
F=EXP(-VAFBQ*TIMF)
FTURN
ENT
```

```
SUBROUTINE BOIRE! (A,N4,DX,RE!)
DIMENSION A(N4)
SUB=0.
X=DX/2.
DC 10 I=1,N4
CALL MTFEC (X,F)
SUB=SUB+A(I)*F*X
X=X+DX
CONTINUE
REL=SUB*LX*6.2831852
BETURN
END
```

```
CC---- IF THE RMS WAVES OF PHASE DISTORTION ABE NOT SPECIFIED,
THEN THIS BOUTINE WILL DETERMINE THE AFFROPRIATE WAVEBO THAT

CC---- PRODUCES THE SPECIFIED TOFLMT.

CCMMON / EC/ TIPLET, VARBO, WAVEEC, SCALEC

DIMENSION A(N4)

RELO=1./TDFLMT**2

VARBO=-ALCG(REIO)

DVEC=VAREC/10.

INSIGNO=1.

CONTINUE

CALL BOIREL (A,N4,DX,REL)

DEEL=REI-BELO / FEIC.LT.TEST) GC TO 40

IF (DREL-IT.0.) GC TC 20

NSIGN1=1

GC TO 30

NSIGN1=-1

GC TO 30

NSIGN1=-1

CONTINUE

IF (NSIGN1.NE.NSIGNO) DVBQ=DVEC/2.

VARBO=VARBO+DVEC**NSIGN1

GC TO 10

CONTINUE

IF (NSIGN1.NE.NSIGNO) DVBQ=DVEC/2.

VARBO=VARBO+DVEC**NSIGN1

GC TO 10

CONTINUE

IF (NSIGN1.NE.NSIGN)

GC TO 10

CONTINUE

IF (NSIGN1.NE.NSIGNO) DVBQ=DVEC/2.

VARBO=VARBO+DVEC**NSIGN1

GC TO 10

CONTINUE

IF (NSIGN1.NE.NSIGNO) DVBQ=DVEC/2.

VARBO-VARBO+DVEC**NSIGN1

GC TO 10

CONTINUE

FETURN

END
```

```
SUBROUTINE JII (A,DX,N4,SIGJII,WAVE)

C---- COMPUTES ARRAY FOR JITTER MTF.

DIMENSION A(N4)
X=DX/2.
DO 10 i= 1,N4
CALL MTFJIT (X,SIGJII,WAVE,F)
A(I)=F
X=X+DX
ETURN
ENC
```

```
SUBROUTINE BY (A, IX, N4)

C---- COMPUTES ARRAY FOR FEAM QUALITY MTF.

DIMENSION A (N4)

X=DX/2
DO 10 I=1,N4

CALL MTFEÇ (X,F)

A (I)=F

10 X=X+DX

EFTURN
END
```

```
SUBROUTINE CN2H (EEIGHT, CN2)

C CACULATE ATMOSFEEEIC VERTICAL TUBBULENCE

C---- HUFNAGELS LATEST CCD2L-GOCD ONLY ABOVE 3 KILOMETERS

CN2=2.2*(1.E-5*10.**(-.3)*HEIGHT)**10.*EXP(-dEIGHT/1000.)

CN2=CN2+1.E-16*EXF(-EEIGHT/1500.)

CN2=CN2*2.7

THIS MODIFICATION IS USED to include turbulence at lower alts

CN2=CN2+1.E-14*EXF(-EEIGHT/100.)

RETURN
END
```

```
C--- COMPUTES THAT FABL OF THERMAL ELCCMING THAT CHANGES WITH SLANT PATH

COMMON /AIMO/ BAIMC
COSSO=COSWND

DELH=(HAIMO+HI)/N

RG=DELH/2.

HEIGHT=HI-DELE/2.

VOCOS=V0*COSSC

SUM=0.

SUMAS=0.
DC 10 I=1,N

CALL ALFA (ALA, HEIGHT)

CALL ALFS (ALS, HEIGHT)

SUMAS=SUMAS+ALS+ALA

ALCSS=EXF(-SUMAS*LEIS/COSCHG)

SUM-SUM+AIOSS*ALA/(VOCOS+RG*PHIDCT)

RG=RG+DELE

HEIGHT=HIGHT+DELH

ESUM**2

RETURN

END
```

```
SUBBOUTINE PRIFES (MI,MLH,WAVF,PH)

C----- COMPUTES THAT FABT OF THERMAL BLOOMING PHASE THAT DOES NOT DEPEND

CN SLANT RANGE

CCMMON / ARBFILD, LIA, DIAGES, BEAMSZ,UO
DIMENSION PH (MIH,ML)
MIHEMELH-1
DIASO-DIA**2

RADSO-DIA**2

RADSO-DIA**2

RADSO-DIA**2

RADSO-DIA**2

RADSO-DIA**2

RADSO-DIA**2

RESO-DIAGES/2.)**2

RISO-DIAGES/2.)**2

RISO-DIAGES/2.)**2

DEIN-DIA/2.**DEIN/2.

YI=-DIA/2.**DEIN/2.

RESCRT (REC)
FACTORO-D.

IF (REC_II.RCSC_AND.FSQ.GT.RISC) CALL FIELD (R,FACTOR)
FACTORO-D.

IF (REC_II.RCSC_AND.FSQ.GT.RISC)

CONTINUE
RESURSON

CONTINUE
RESU
```

```
SUBROUTINE RESEBS (MI,MLH,PH)
             CALCULATES THE RESIDUAL PHASE AFTER THE MEAN, TILT, AND FOCUS HAVE BEEN REMOVED.
              DIMENSION PH(MIH, MI)
COMMON /ARBPLE/ LIA, DIAOBS, BEAKSZ, UO
MIHM-MIH-1
MIE-MI-1
DEIN-DIA/MIM
XI-DEIN/2.
YI--DIA/2.+DEIN/2.
RCSQ=(DIA/2.) **2
BISQ=(LIACBS/2.) **2
000
                      COMPUTE NOREALIZATION CONSTANTS
              CAIL COEFS (AC, A1, A2, A3, A4, MIM, MLHM, DELN)
             COMPUTE EXPANSION COEFFICIENTS
              SUM1=0.
SUM2=0.
SUM3=0.
SUM4=0.
X=XI
DO 20 I=1,MLHM
XSC=X**2
Y=YI
DC 10 J=1,MLM
YSC=Y**2
RSC=XSC+YSC
IF (RSC.GI.RCSC.OE.RSC.LT.RISC) GO TC 10
CALL FIELD (SCFI (FSC),UR)
     --- DETERMINE EXPANSION COEFFICIENTS RELATIVE TO A UNIFORM --- WEIGHTING FUNCTION.
              FI=0.
IF (ABS (UR) - GI. 1. F-2) PT= 1.
FACTOR=PE(I,J) *FI
SUM 1=SUM 1+FACICE*X
SUM2=SUM2+FACICE*X
SUM3=SUM3+FACICE*ESQ
SUM4=SUM4+FACICE
Y=Y+DELN
X=X+DELN
--- ADJUST FOR HALF PLANE INTEGRATION
SUM1=0.
SUM1=0.
SUM2=SUM2*2.
SUM3=SUM3*2.
SUM4=SUM4*2.
10
20
C-
C
               DEINSQ=DEIN**2
PHMEAN=AC*SUM4*DEINSC
PHILTX=A1*SUM1*DEINSC
PHILTY=A2*SUM2*DEINSC
PHFOC=(A3*SUM3+A4*SUM4)*DELNSC
2000
       --- SUBTRACT THE MEAN, TIIT, AND FOCUS CURVATURES.
            X=XI

DC 40 I=1,MLHM

XSC=X**2

Y=YI

DO 30 J=1,MLM

YSQ=Y**2

BSC=XSC+YSQ

IF (RSO_GI.ROSC.OR.RSO.LT.RISC) GO TO 30

PH(I,J)=FH(I,J)-FHTLIX*A1*X-PETLTY*A2*Y-PHFOC* (A3*RSQ+A4)-PHMEAN*A

10

Y=Y+DELN

X=X+DELN

X=X+DELN

EFTURN

END
30
```

```
C---- COMPUTES TILT AND FOCUS NORMALIZATION CONSTANTS

COMMON / REPILL DIA, DIAOPS, BEAMSZ, UP
ROSSO = (DIACES, 2.) **2
ROSSO = (DIACES, 2.) **2
PIENSO = 2118*2

ROSSO = (DIACES, 2.) **2
PIENSO = 2118*2

VI = DIA V = 1 **

NUM = 0.

DO 20 I = 1, MLH

VSC = V = 2

DO 10 J = 1, MLH

VSC = V = 2

LF RSSO CI.EOSC.OE.RSO.LT.RISC) GO TO 10

COMMON FIRE PLANT NORMALIZATION COEFFICIENTS RELATIVE TO A UNIFORM

COMPONENT NORMALIZATION COEFFICIENTS RELATIVE TO A UNIFORM

PT = 0.

FI (ABS (UB) .GT .1.E-2) FT = 1.

FACTOR = FT

SUM = SUM + FACTOR **
SUM = S
```

```
C----COMPUTES THE THERMAI BLOOMING FHASE VARIANCE
C----VARIANCE IS CCEPUTED LIKE THE STREHL RATIO ACCORDING TO THE
C----FIELD AS A WEIGHTING FUNCTION.

COMMON / ARBFLL IIA, DIAOBS, BEAMSZ, UO

DIMENSION PHOFILE, ELD

RCSQ= DIA / 2. *** **2

RLSQ= DIA / 2. *** **2

MINEHLT-1

DELN-DIA / 2. ** **1

MINEHLT-1

DELN-DIA / 2. ** **X

X = XI

SUM 1= 0.

SUM 2= 0.

SUM 3= 0.

SUM 3= 0.

SUM 3= 0.

SUM 3= 0.

CAIL PIELD (SCFT (FSC), UR)

SUM 1= SUM 1+UR* FE (I, J) **2

SUM 3= SUM 3+UR* FE (I, J) **2

SUM 3= SUM 3+UR* FE (I, J) **2

CONTINUE

20 CONTINUE

21 CONTINUE

22 SUM 3= SUM 1+2. ** ELIN **2

SUM 3= SUM 1+2. ** ELIN **2

SUM 3= SUM 1+2. ** ELIN **2

SUM 3= SUM 3+2. ** ELIN **2

CC

RETURN

END
```

```
SUEROUTINE ISCIRB (N, HT, HG, W, ISCANG)

C---- COMPUTE ZENITH ISCFLANATIC ANGLE BASED UPON D.L. FRIEDS

C---- DEFINITION

COMMON /AIMO/ EAIMO

REAL ISCANG

IPI=4.*ARSIN(1.)

CK=TPI/W

CKSQ=CK*CK

DEIH=(HAIMO-HI)/N

HEIGHT=HG+DELH/2.

SUM=0.

DC 10 1=1, N

CALL CN2E (HEIGHT CN2)

SUM=SUM+CN2*DHT**1.66666667

HEIGHT=HHIGHT+LELH

CCNTINUE

CN1=SUM*FILH

LCN1=SUM*FILH

LEN=2.91*CKSC*CN1

ISOANG=(6.88/LEN)**.6*.314

RETURN
```

```
SUBROUTINE ISCFIA (JC,XNU,R,SECCMG,WAVE,XINT,HT,HG,TISO,N4,II)
C ---- THIS SUBROUTINE COMPUTES THE ISOPLANATIC MTF FOR A FULL AO C ---- COMPENSATED SYSTEM DEVELOPED BY DL PRIED
                                               -THIS SUBROUTINE CCHEUTES THE ISOPLANATIC NOT A FULL AO
-COMPENSATED SYSTEP DEVELORED BY DL PRIED
DIENSION H23) (A (19), C2 (19), C4 (19)
DIENSION H23) (A (10) (20), CN2 (23)
TO CREATE THE PRINCE OF THE PRIED
TO CREATE THE PRIED OF TO CREATE THE PRIED
THE HAT NO E I / NE
BELL CREATE THE HEAT HOLD BY THE HEAT HO
   10
   40
   50
   60
   70
    80
    90
100
110
    120
C
C
C
                                                                           NOW INTEGRATE CYEE PHI
                                                       CSI=R**A53*SECCMG/WAVE**2
SUM=0.
DC 130 I=1,N
SUM=SUM+EXP(-XMU(I+1)*CST)
    130
C
C
```

XINT=SUM*DPHI*2./FI TISO(II) = XINT RETURN END

С

```
C----
THIS SUBROUTINE DETERMINES WHAT VALUE OF RHO ROULD PRODUCE THE AC CORRECTED STREEL AS IF THERE WERE NO COMPENSATION AT ALL

REAL IRRETF(N4)

REBHOU DESEROU/2.
NSIGNO=-1
NSIGN=-1
CONTINUE CALL TREEL (IFRET, CLA, N4, DX, SE, R)
IF (ABS(SF-EXD)/FXD.LT..1) GG TO 40
DIFSR-EXI
IF (DIF.11.0.) GC TC 20
NSIGN=-1
GC TO 30
NSIGN=-1
30 CONTINUE
IF (DIF.GE.O.) DE=DE/2
NSIGNO=NSIGN
EF-DR*NSIGN
GC TO 10
CONTINUE
HGC-E
RETURN
END
```

```
SUBROUTINE TREREL (IRRMTF,DIA,N4,DX,TREI,RHO)

C---- THIS SUBROUTINE COMPUTES THE RELATIVE INTENSITY DUE TO A TOTAL

BEAL IRRMTF (N4)
PI=2.*ARSIN(1.)
SUM=0.
CY2=DX/2.
X=DX2
DC 10 I=1,N4
F1=IRRMTF(I)
CALL MITF(I)
SUM=SUN+F1*F2*1
X=X+DX
X=X+DX
CONTINUE
TREL=SUM*2.*FI*DX
RETURN
END
```

```
SUBROUTINE TLIISC (D,OMX,OMY, BESTLT, SECONG, HTE ANS, HABVGD)
C C----THIS SUBROUTINE COMPUTES THE BESIDUAL TILT DUE TO ISOPLANATISM C
       COMMON /ATMO/ BATMO

N=100

E56=5./6-

E53=5./3-

HATMC-HTRAKS)/N

H=BABVGD+LS/2

SUB=0.
С
       DO 10 I=1,N
С
       CALL CN2E (H,Ch2)
С
       Z=SECOMG*H
С
       С
       SUM=SUM+ (4.*AEG4+4.*AEG3-AEG2-AEG5-AEG1-AEG6) *CN2 H=H+DS CONTINUE
10
C
C
   -- THIS IS THE RESULTANT 2-SIGMA-P TILT DUE TO ISOPLANATISM.
       RESTLT = SUM*DS*SECCMG*2.91/D**2*2.
C
       SETURN
END
```

```
SUBROUTINE RELICI (1,TR,TM)

C ---- WE HAVE FOUND TEE DETAILED MIF CODE TO GIVE RESULTS

C---- WHICH LIE BEIWEEN AN RSS MODEL FOR ELCOMING AND A

C---- MULTIPLICATIVE AFFRCACH. THIS SUBROUTINE IS AN ATTEMPT

C---- AT BETTEE MATCHING THE RESULTS OF TEE MTF CODE BY

AVERAGING THE ESS AND THE MULTIPLICATIVE RESULTS.

C

T=(TR+TM)/2.

BETURN
END
```

```
SUBROUTINE BLCCM (D,E,S,T)

C ----THESE BLCCMING IREI MCDELS ARE CURVE FITTED TO GUISMTF
C----BUNS FOR SIGS2'S GREATER THAN APOUT 1.2. THE APERTURE
C----DISTRIBUTIONS USEL HAD OBSCURATIONS OF .1 X THE OUTER
C----TO THE GUISMTF CODE AGAIN.

SCRIS-SORT(S)
TRG=-.08705+SQRIS*2.91485+.1723*S
TRG=1./TRG
IEU=1./TRG
TRU=1./TRU
T=(1.- (D/E) **2)*TRU+(D/B) **2*TRG
IE (1.- (D/E) **2)*TRU+(D/B) **2*TRG
IE (D.GI.E) T= TRG
RETURN
END
```

```
SUBROUTINE SCINT (SIGKZ, N, HTRANS, HGRND, WAVE)

C---- THIS ROUTINE COMEDTES THE VABIANCE OF TEE LOG AMPLITUDE
C---- POR A ZENITH ANGLE. THE RESULTS WILL ONLY BE USED

IN THE EVENT TEAT FUIL AO IS CTILIZED IN THE RUN.

COMMON /AIMO/ EATMO
CK=6.28/AVE
HIOTAL=HAIMO-HIFAKS
DH=HTOTAL/N
H=HTRANS+DH/2.
C56=5./6.
SUM=0.
DC 10 I=1, N
H56=H*C56
CALL CN2E (H, CN2)
SUM=SUM+CK2*H56
H=B+DH
CCNTINUE
CK76=CK**(7./6-)*LH*C.56
RETURN
END
```

SUBROUTINE SINIIS (TAMP, SIGXZ, SECOMG)

C ---- COMPUTE THE LOG AMPLITUDE SCINIILLATION VARIANCE
C---- FOR OFF-ZENITH CCADITIONS AND THE RELATIVE IRRADIANCE
C---- REDUCTION WHEN FUIL AO IS USED.

TAMP=EXP (-SIGXZ*SECCMG** (11./6.))
RETURN
END

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